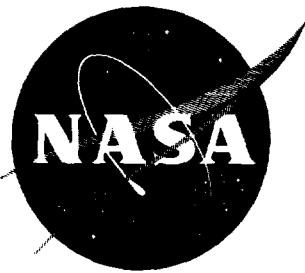


35p.

76 N62-16838

NASA TN D-1429

NASA TN D-1429



TECHNICAL NOTE

D-1429

THERMODYNAMIC AND TRANSPORT PROPERTY CORRELATION

FORMULAS FOR EQUILIBRIUM AIR

FROM 1,000° K TO 15,000° K

By John R. Viegas and John T. Howe

Ames Research Center
Moffett Field, Calif.

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

WASHINGTON

October 1962

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

TECHNICAL NOTE D-1429

THERMODYNAMIC AND TRANSPORT PROPERTY CORRELATION

FORMULAS FOR EQUILIBRIUM AIR

FROM 1,000° K TO 15,000° K

By John R. Viegas and John T. Howe

SUMMARY

The thermodynamic properties, density and temperature, as well as transport property parameters involving viscosity, Prandtl number (including diffusion effects), and gaseous radiation absorption coefficients have been correlated as a function of enthalpy at four pressure levels (10^{-1} , 10^0 , 10^1 , and 10^2 atmospheres). The correlation formulas are written in a generalized form for which coefficients for a particular property and pressure level are tabulated. The correlation formulas are useful in digital computer programs for nonadiabatic viscous flow problems.

INTRODUCTION

Thermodynamic and transport properties of high temperature air as well as their derivatives with respect to enthalpy at constant pressure are often needed for the computation of flow fields on bodies in high-speed flight. These properties are available in tabular form (refs. 1 and 2). However, this form is often not very convenient for use in digital computers. To facilitate machine computations, it is sometimes faster and easier if the properties are represented by analytical expressions which can also be readily differentiated. Cohen (ref. 3) correlates density and some transport properties independently of pressure for flight speeds up to 29,000 ft/sec. Correlations are now required for high speeds up to 50,000 ft/sec to facilitate studies of high-speed entry into the earth's atmosphere.

At these higher speeds, in excess of approximately 30,000 feet per second, the transport properties of air are significantly affected by ionization. Furthermore, at these speeds, gaseous radiation effects can also be important, depending on body size and altitude (ref. 4). Thus, for entry into the earth's atmosphere on return from the moon, the planets, or far out in the solar system, for which entry speeds will be between 35,000 and 50,000 feet per second, both ionization and radiation effects may be important and should be considered. For these reasons, thermodynamic and transport properties of equilibrium air at temperatures up

to $15,000^{\circ}$ K (stagnation temperature for flight at 50,000 ft/sec at approximately 190,000 ft altitude) as presented in references 1 and 2, are correlated as functions of enthalpy at the four pressure levels, 10^{-1} , 10^0 , 10^1 , 10^2 atmospheres in the present work. The Planck mean mass absorption coefficient for gaseous radiation, as presented by both references 5 and 6, is also correlated as functions of enthalpy at the same pressure levels.

SYMBOLS

a,b,c,d,e	constant and coefficients in equation (1)
h	enthalpy, ft^2/sec^2
k	Planck mean mass absorption coefficient, ft^2/slug
p	pressure, atm
Pr	Prandtl number
T	temperature, $^{\circ}\text{K}$
x	independent variable in equation (1), $\frac{h}{h_r}$
y	dependent variable in equation (1) (appropriate property)
ρ	mass density, slug/ft^3
μ	viscosity coefficient, $\text{lb sec}/\text{ft}^2$

Subscripts

a	results obtained from reference 5
l	results obtained from reference 6
1,2, . . . n	e coefficients in equation (1)
r	reference conditions in table II.

CORRELATION FORMULAS

Many attempts were made to fit smooth curves through the desired property values obtained from references 1, 2, 5, and 6. Polynomials of all degrees up to 8 with coefficients determined by the method of least squares and the method of Tchebycheff were tried. Fourier series and generalized conics were also tried. In some cases it was found necessary to join as many as four sections of curves smoothly in series to obtain adequate correlations. An attempt was made to minimize possible discontinuities at the joints in segmented curves either by overlapping the sections and choosing a suitable point in this overlapped region to be the limit of the various curves (this was done for conics), or by matching the slope and property value of two adjoining curves at the same value of enthalpy (this was done for polynomials). The curves presented in this paper are the best results, from the methods attempted, for obtaining accurate and smoothly varying property values as functions of enthalpy.

Although various methods of correlation are used, it is convenient to express the correlation of all properties at a specific pressure level by the general formula

$$a + by + cxy + dy^2 + e_1x + e_2x^2 + e_3x^3 \dots e_nx^n = 0 \quad (1)$$

where the independent variable x is the enthalpy ratio h/h_r and the dependent variable y is the appropriate gas property. It is seen that if the coefficients $e_3 \dots e_n$ are zero, the equation is that of a general conic with inclined axis, and if the coefficients c and d are zero, the equation is that of a polynomial of degree n .

To facilitate the use of equation (1), the coefficients for the various properties at the pressure levels considered are presented in table I. This table shows the type of correlating function used (general conic or polynomial of degree n) for each property, the upper and lower enthalpy limits for the validity of each section of the correlation curve, and, for those properties fit by a general conic, the sign of the appropriate root is also given.

The over-all limits of validity of these correlation formulas for each property correspond to a temperature range of approximately $1,000^\circ$ to $15,000^\circ$ K. For convenience, each property is referenced to a standard condition. The pressure is referenced to sea-level conditions. The reference conditions of all other properties correspond to their values at satellite enthalpy ($h_r = 3.125 \times 10^8 \text{ ft}^2/\text{sec}^2$ or $12,474 \text{ Btu/lb}$) at each pressure level. The reference conditions are listed in table II.

DISCUSSION OF RESULTS

The thermodynamic and transport properties as obtained from the correlation formulas presented in this report and the properties they represent from references 1, 2, 5, and 6 are compared in figures 1 through 5. In general, the agreement is good. The analytical expressions should provide property values with sufficient accuracy for most machine calculations. In the remainder of this discussion, consideration is given to certain features of the correlations.

In figure 3 where the ratio of the density-viscosity product to the Prandtl number is correlated, no attempt was made to fit the minor variation in the property values which occurred at low enthalpies (near $h/h_r = 0.3$). It was felt that the effect of this variation would be negligible in comparison to the effects of the over-all variation. The peak in each curve corresponds to the onset of ionization. The correlation is seen to be very good in the ionization regime.

To study the effects of energy transport by gaseous radiation, the Planck mean mass absorption coefficient is useful. It has been calculated from theory in reference 6 and has been obtained by a combination of theory and experiment in reference 5. The two references are in reasonable, but not close, agreement. The results of both are correlated in figure 5.

First the absorption coefficient of reference 6 as correlated for all pressure levels is shown in figure 5(a). The fit is fair except for the point at $h/hr \approx 3.9$ for 1 atmosphere pressure. The high point at each pressure level corresponds to $15,000^\circ K$ and was obtained by a graphical and logarithmic interpolation of results in reference 6 at $12,000^\circ$ and $18,000^\circ K$.

The correlation of the absorption coefficient at individual pressure levels is shown in figures 5(b) through 5(e) corresponding to the results of reference 6 and figure 5(f) through 5(i) corresponding to the results of reference 5. Except for figure 5(f), the correlation is satisfactory. No attempt was made to fit figure 5(f) because of the lack of a point defining the middle range of the properties.

Finally, it is instructive to go back and examine the thermal conductivity used in the Prandtl number of figure 3. This is especially pertinent because the current lack of agreement between the stagnation point convective heat-transfer rates in references 7 and 8 may be attributed to the thermal conductivity of ionized air.

In the present paper, the thermal conductivity used in the Prandtl number includes the effects of energy transfer by both molecular collisions and diffusion of molecular species (ref. 2). The thermal conductivity calculated for air by Hansen (ref. 2) agrees quite well with experimental results at temperatures up to $5,000^\circ K$ as reported by Peng and Ahtye

(ref. 9). At higher temperatures, Hansen's results can be compared with the conductivity deduced experimentally for nitrogen by Maecker (ref. 10). In this comparison, shown in figure 6, agreement is fairly good and the relative magnitudes of conductivities of the two gases are as expected (see refs. 9 and 11). Results of King (ref. 12) for the thermal conductivity of pure nitrogen agree very well with those of Maecker.

CONCLUDING REMARKS

Thermodynamic properties and transport property parameters have been correlated as a function of enthalpy at four pressure levels. In general, the correlation formulas represent the properties quite accurately. Although the formulas are lengthy, they can be evaluated very rapidly by digital computers. For example, a property represented by an eighth-degree polynomial can be evaluated at 1000 points in approximately 0.7 second on an IBM 7090 data processing machine. This is one to two orders of magnitude faster than having this machine look up the same number of points in a table. Thus, the formulas are expected to be useful for flow-field computation on digital computers.

Ames Research Center

National Aeronautics and Space Administration
Moffett Field, Calif., June 21, 1962

REFERENCES

1. Moeckel, W. E., and Weston, Kenneth C.: Composition and Thermodynamic Properties of Air in Chemical Equilibrium. NACA TN 4265, 1958.
2. Hansen, C. Frederick: Approximations for the Thermodynamic and Transport Properties of High-Temperature Air. NASA TR R-50, 1959.
3. Cohen, Nathaniel B.: Correlation Formulas and Tables of Density and Some Transport Properties of Equilibrium Dissociating Air for Use in Solutions of the Boundary-Layer Equations. NASA TN D-194, 1960.
4. Yoshikawa, Kenneth K., and Wick, Bradford H.: Radiative Heat Transfer During Atmosphere Entry at Parabolic Velocity. NASA TN D-1074, 1961.
5. Kivel, B., and Bailey, K.: Tables of Radiation From High Temperature Air. Avco-Everett Research Laboratory. Res. Rep. 21, 1957.
6. Armstrong, B. H., Sokoloff, J., Nicholls, R. W., Holland, D. H., and Meyerott, R. E.: Radiative Properties of High Temperature Air. Journal of Quantitative Spectroscopy & Radiative Transfer, vol. 1, no. 2, Nov. 1961, pp. 143-162.
7. Scala, Sinclair M., and Warren, Walter R.: Hypervelocity Stagnation Point Heat Transfer. General Electric, Space Sciences Laboratory, R61SD185, 1961.
8. Pallone, Adrian, and Van Tassel, William: Stagnation Point Heat Transfer for Air in the Ionization Regime. ARS Jour., vol. 32, no. 3, Mar. 1962, p. 436.
9. Peng, Tzy-Cheng, and Ahtye, Warren F.: Experimental and Theoretical Study of Heat Conduction for Air Up to 5000° K. NASA TN D-687, 1961.
10. Maecker, H.: Thermal and Electrical Conductivity of Nitrogen Up to 15,000° K by Arc Measurements. Presented at Meeting on Properties of Gases at High Temperature, AGARD, Aachen, Sept. 21-23, 1959.
11. Ahtye, W. F., and Peng, Tzy-Cheng: Approximations for the Thermodynamic and Transport Properties of High-Temperature Nitrogen With Shock-Tube Applications. NASA TN D-1303, 1962.
12. King, L. A.: Theoretical Calculation of Arc Temperatures in Different Gases. Colloquium Spectroscopicum Internationale VI, (Amsterdam, 1956), pp. 152-161 (London: Pergamon Press, 1957). (Based on E.R.A. Report Ref. G/XT-155).

TABLE I.- COEFFICIENTS

Property, γ	Pressure level, atm	Type of curve	x limits for curve		a	b	c	d	e_1	e_2	e_3	e_4	e_5	e_6	e_7	e_8	Root of quadratic
			Lower	Upper													
$\frac{Pr}{P}$	10^{-1}	(a)	0.03608	5.4273	0.06592768	-1	0	0	1.505904	-1.165208	0.9343309	-0.3375398	0.05518693	-0.003330603	0	0	
	10^0	(a)		3.9643	.007398195	-1			2.109765	-2.739214	2.364143	-1.8956323	.1007640	.009547003	-.002160125	0	
	10^1	(b)		2.6736	.01948670	-1			1.829233	-2.967630	7.119123	-11.11281	9.714190	-4.594607	1.105638	-.1064150	
	10^2	(a)		2.1589	.01752455	-1			1.533830	-9.9134607	.5619979	-3.583186	.1964153	-.04129094	0	0	
$\frac{\alpha_1}{D_T^{1.5} T}$	10^{-1}	(c)		5.4273	7.999610	-5.624454	-2.135941	1	-1.005803	-.04987680	0	0	0	0			-
	10^0			3.9643	2.967720	-4.335691	12.84509	1	-15.59082	3.058515							+
	10^1			2.6736	-13.99580	-1.443177	89.74126	1	-89.68862	15.39430							+
	10^2			2.1589	.1344780	-4.118392	28.04161	1	-30.05294	4.841396							+
$\frac{\alpha_1 P_T}{D_T^{1.5} T^2}$	10^{-1}			1.4	15.16116	-7.702873	2.738838	1	-19.90980	7.949521							-
				1.4	1.6	-.9376820	1	-.5666152	0	.8101086	-.2057187						+
				1.6	1.835	25.02171	-9.583479	3.787318	1	-19.55076	3.657150						+
				1.835	5.4273	-5.528138	1	1.029237	0	.0006060	.06293130						+
	10^0			.03608	.45	-4.674994	1	12.72930	0	-19.98398	9.350414						-
				.45	1.6	108.2544	-45.80435	23.90831	1	-130.1215	42.30918						-
				1.55	2.35	-33.45746	8.832652	-7.821374	1	30.56290	4.802969						-
				2.35	3.9643	-3.994343	1	.3691629	0	.9779776	-.03073380						-
				.03608	1.0	14.04183	-7.002489	-24.87600	1	41.25561	-25.31520						+
				.80	1.64	-172.8071	72.84828	-39.60441	1	203.8301	-64.21053						-
				1.6	2.6736	-17.96245	6.135884	-5.637358	1	15.14080	-1.694240						-
				.03608	.68	8.065993	-6.511189	50.27557	1	-109.8963	59.75427						+
				.68	1.5	-2.468844	1	.3340554	0	1.417372	-.3146546						-
				1.5	1.81	-24.14852	16.07877	-9.454224	1	22.63899	-5.168076						+
				1.78	2.16	-17.93436	9.928399	-7.072455	1	11.35373	-.1983110						-
	10^{-1}	(a)		.03608	.08	-.07161016	-1	0	0	7.595930	-35.45260						
				.08	5.4273	.1364769	-1			2.411026	-3.378803	2.901134	-1.302484	.3106245	-.03751842	.0018102816	
	10^0			.03608	2.54	.01726946	-1			3.500558	-6.042997	5.382014	-2.327105	.4793450	-.03780220	0	
	10^1	(a)		.03608	2.6736	.05531397	-1			.2121646	0	0	0	0	0		
	10^2	(a)		2.1589	.03401435	-1			2.455306	-2.249730	.2606849	.8472206	-.4429157	.06444144			
$\frac{T}{T_T}$	10^{-1}	(c)		5.4273	0	68.85393	-33.18305	1	-37.97243	6.520246	0	0				+	
	10^0			3.9643		28.01277	-18.74220	1	-15.02563	7.682671						+	
	10^1			2.6736		133.9065	-51.91449	1	-27.90829	49.49101						+	
	10^2			2.1589		23.19744	-19.39758	1	-12.22303	7.615397						+	
All levels				5.4273		75.18458	-33.43958	1	-24.53630	-13.04419						+	
$\frac{k}{K_{T/a}}$	10^{-1}	(c)		5.4273		0	0	0	0								+
	10^0			3.9643		49.33526	-35.56380	1	-15.28951	8.139395						+	
	10^1			2.6736		80.99429	-39.79852	1	1.248875	-39.13508						+	
	10^2			2.1589		1634.040	-689.9332	1	-305.7013	-234.0838						+	

^aPolynomial (least squares)^bPolynomial (Tchebycheff)^cConic

TABLE II.- REFERENCE CONDITIONS

Pressure level, atm	h_r , ft^2/sec^2	ρ_r , slug/ft^3	$\rho_r \mu_r$, $\text{lb}^2 \text{ sec}^3/\text{ft}^6$	$\frac{\Pr_r}{\rho_r \mu_r}$, $\text{ft}^6/\text{lb}^2 \text{ sec}^3$	T_r , $^\circ\text{K}$	$(k_r)_l$, ft^2/slug	$(k_r)_a$, ft^2/slug
10^{-1}	3.125×10^8	0.6271×10^{-5}	1.8254×10^{-11}	0.4777×10^{11}	6400	1.66×10^1	2.42×10^1
10^0	3.125×10^8	$.5700 \times 10^{-4}$	1.8065×10^{-10}	$.4694 \times 10^{10}$	7200	6.68×10^1	9.68×10^1
10^1	3.125×10^8	$.5185 \times 10^{-3}$	1.7754×10^{-9}	$.4613 \times 10^9$	8150	2.11×10^2	3.87×10^2
10^2	3.125×10^8	$.4697 \times 10^{-2}$	1.7556×10^{-8}	$.4460 \times 10^8$	9350	8.62×10^2	1.82×10^3

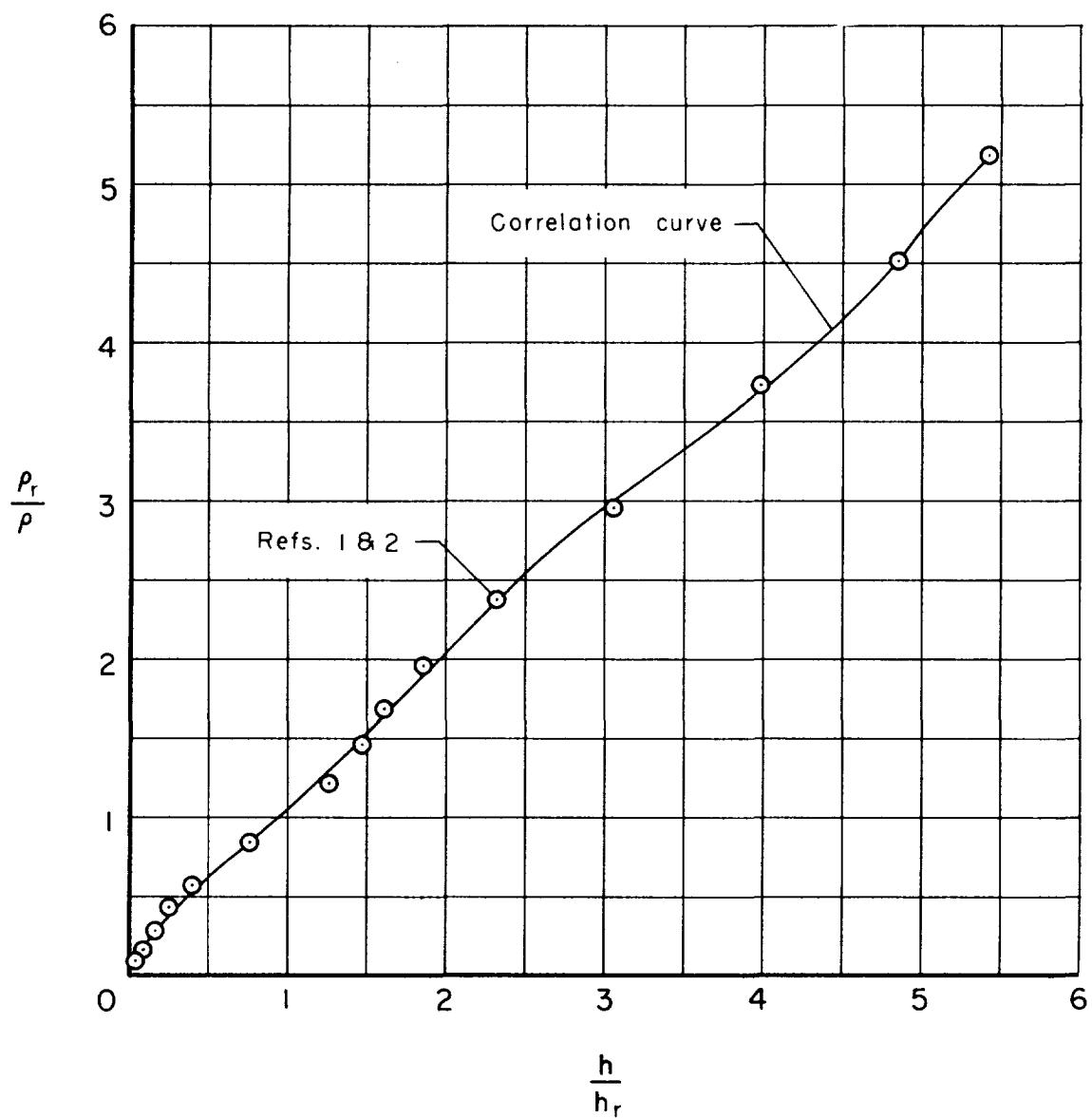
(a) $p = 10^{-1} \text{ atm}$

Figure 1.- Density correlation.

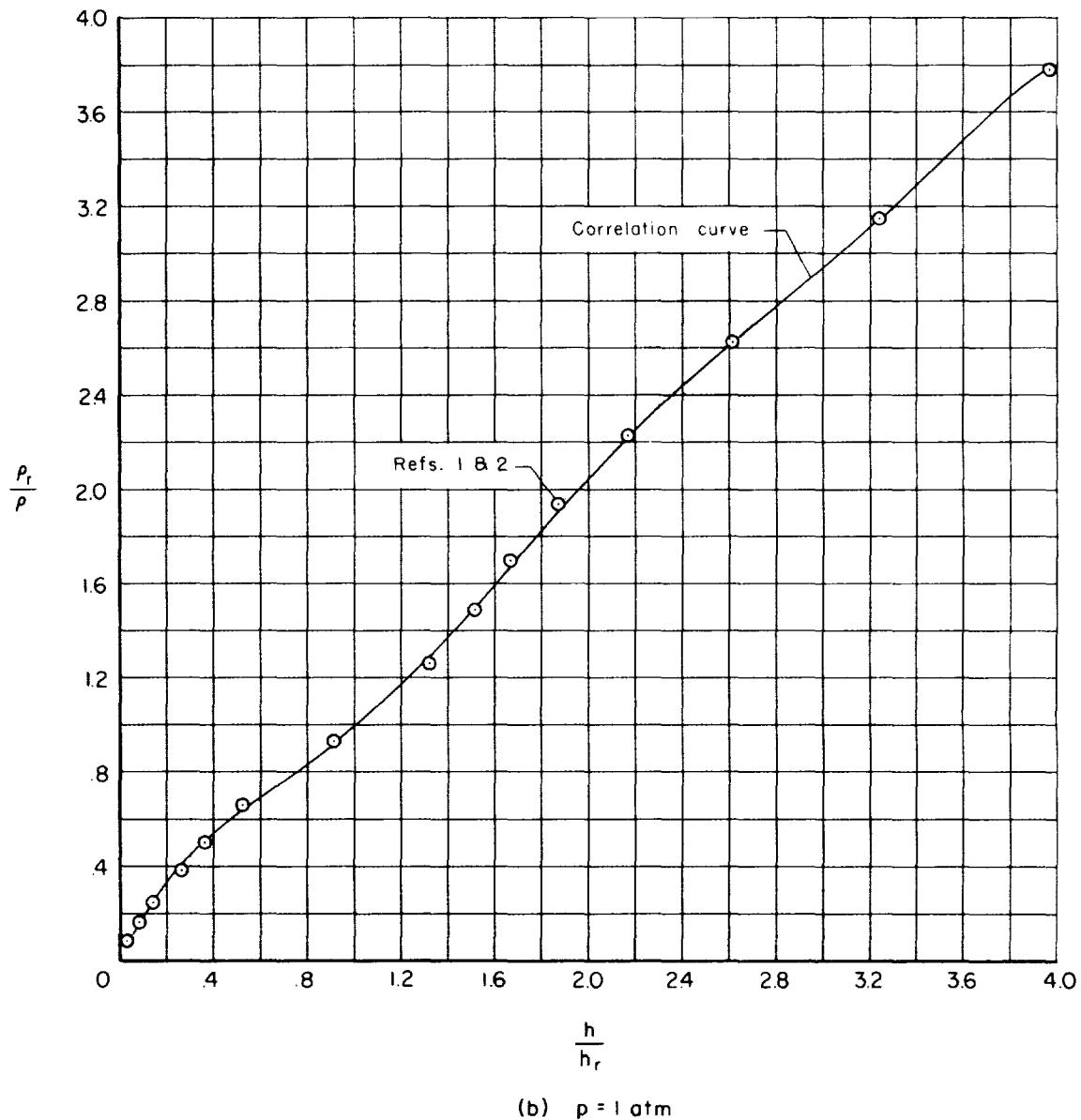


Figure 1.- Continued.

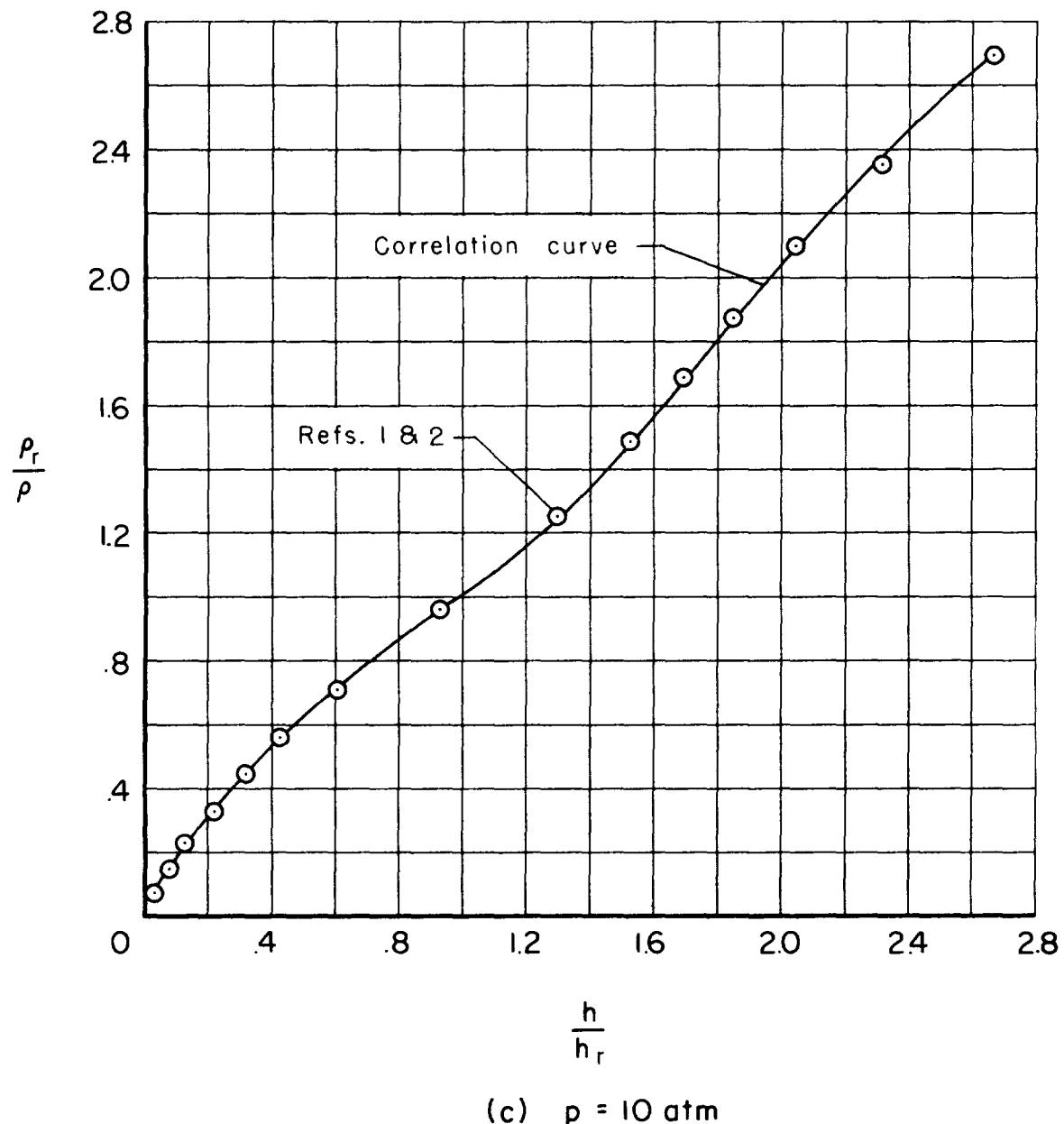


Figure 1.- Continued.

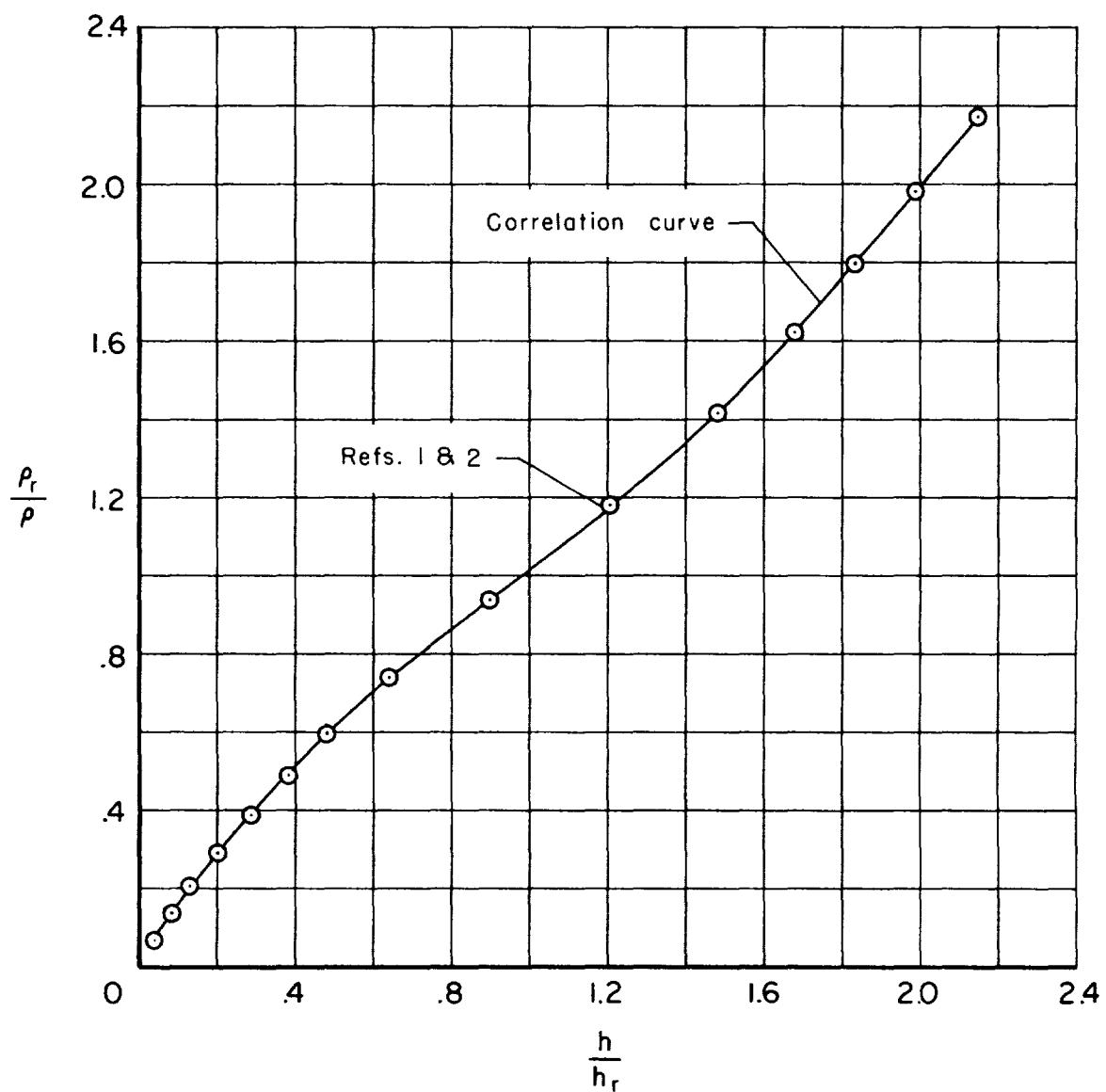
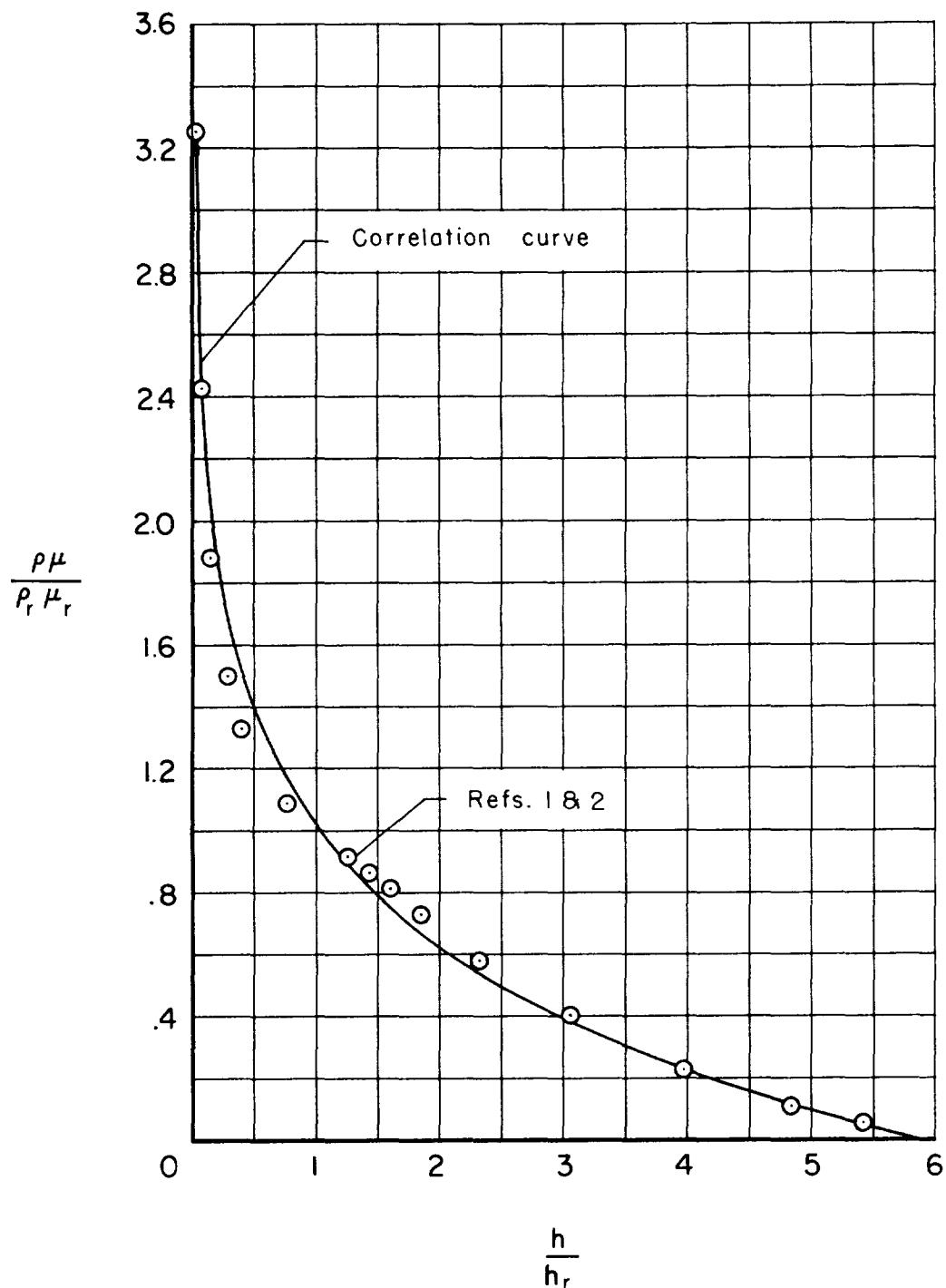
(d) $p = 100$ atm

Figure 1.- Concluded.



(a) $p = 10^{-1} \text{ atm}$

Figure 2.- Density-viscosity parameter correlation.

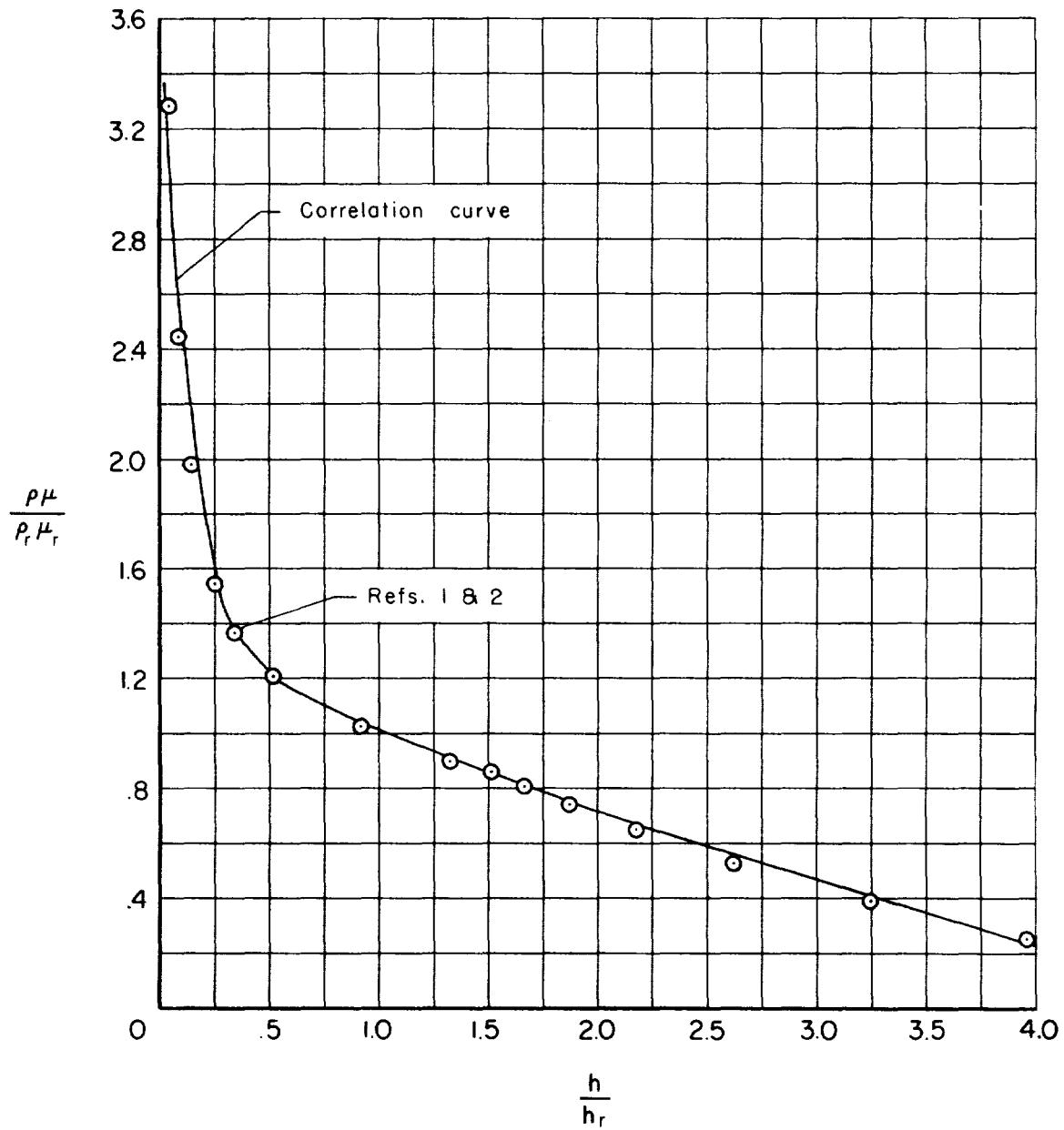
(b) $p = 1 \text{ atm}$

Figure 2.- Continued.

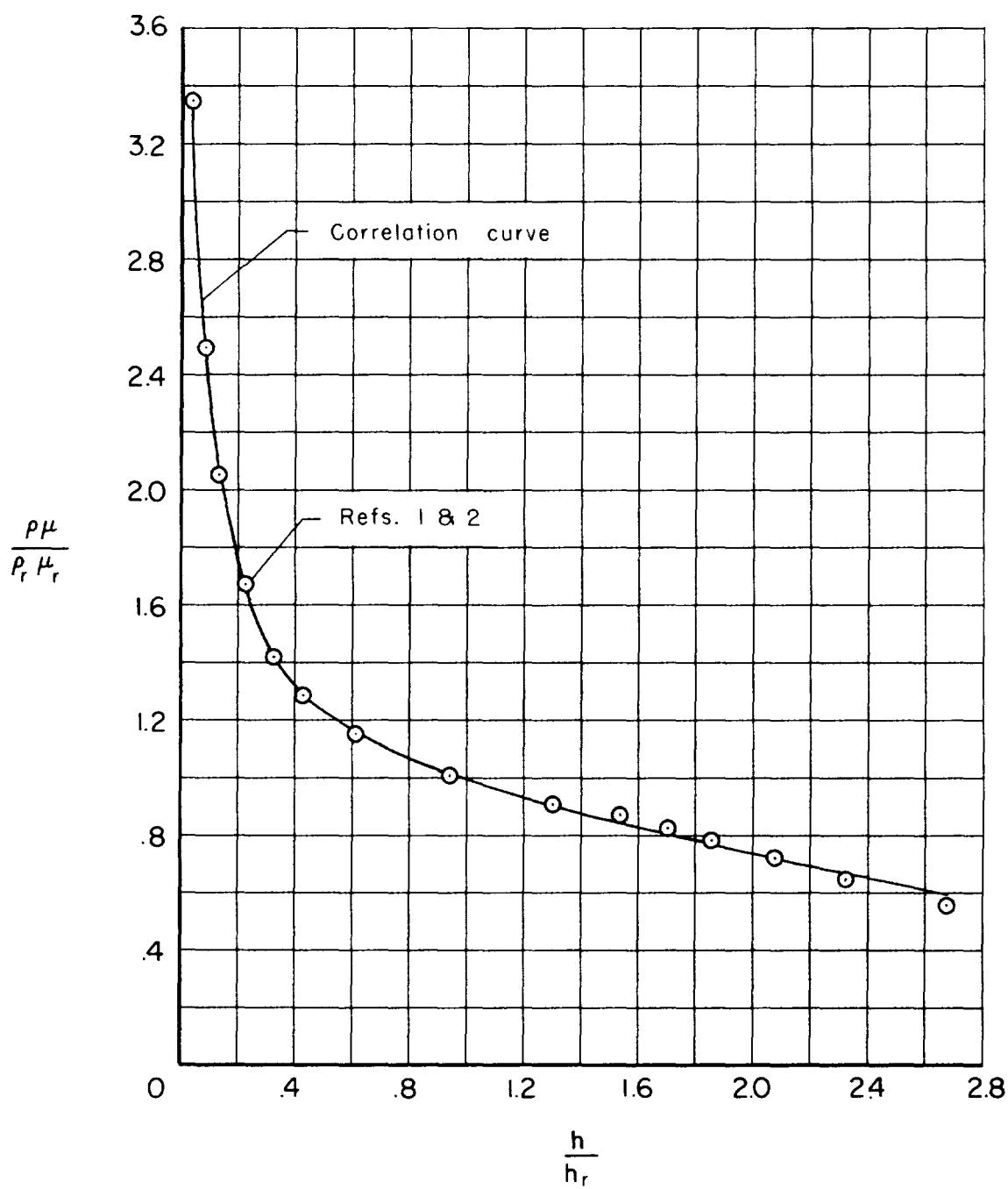
(c) $p = 10 \text{ atm}$

Figure 2.- Continued.

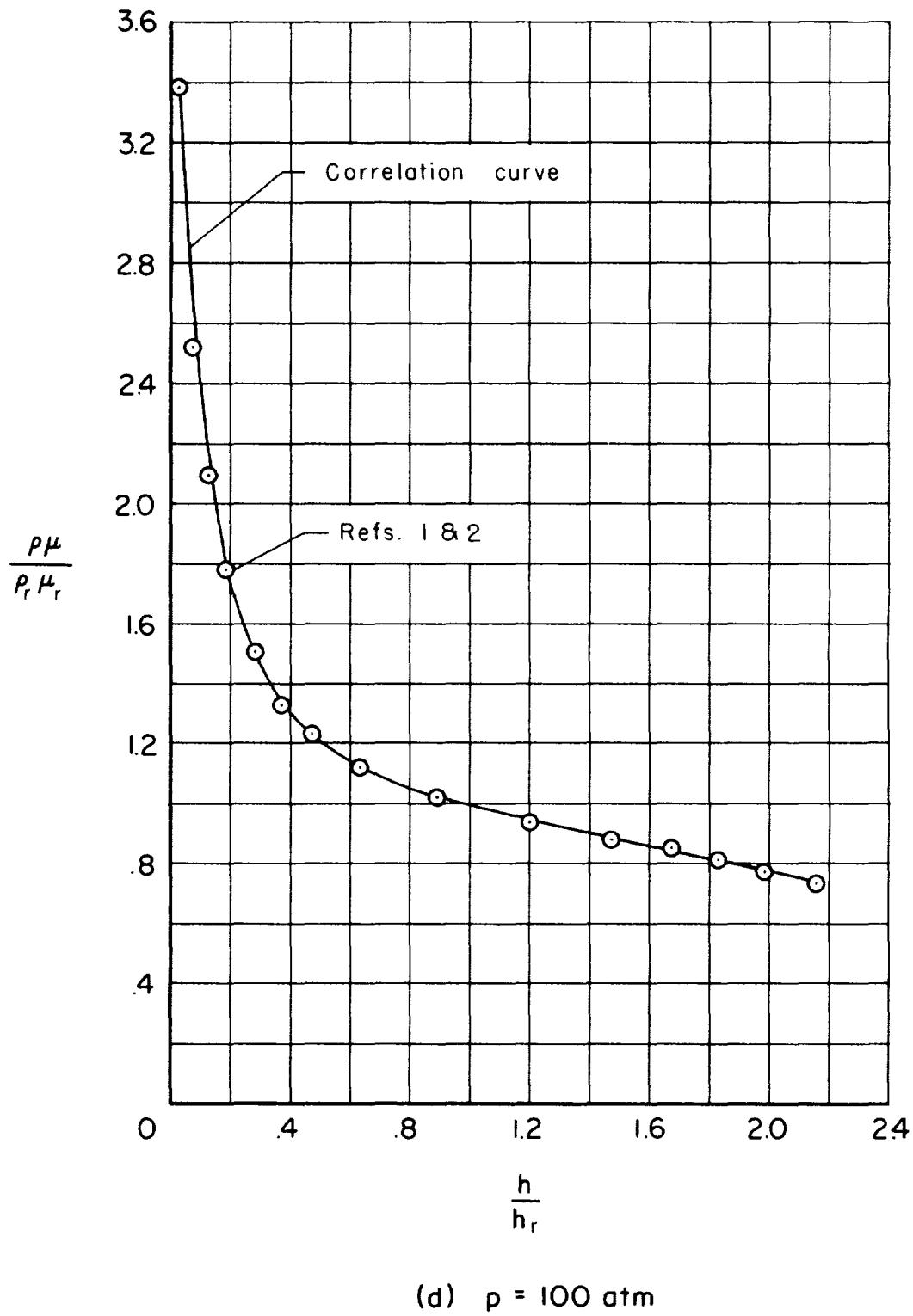


Figure 2.- Concluded.

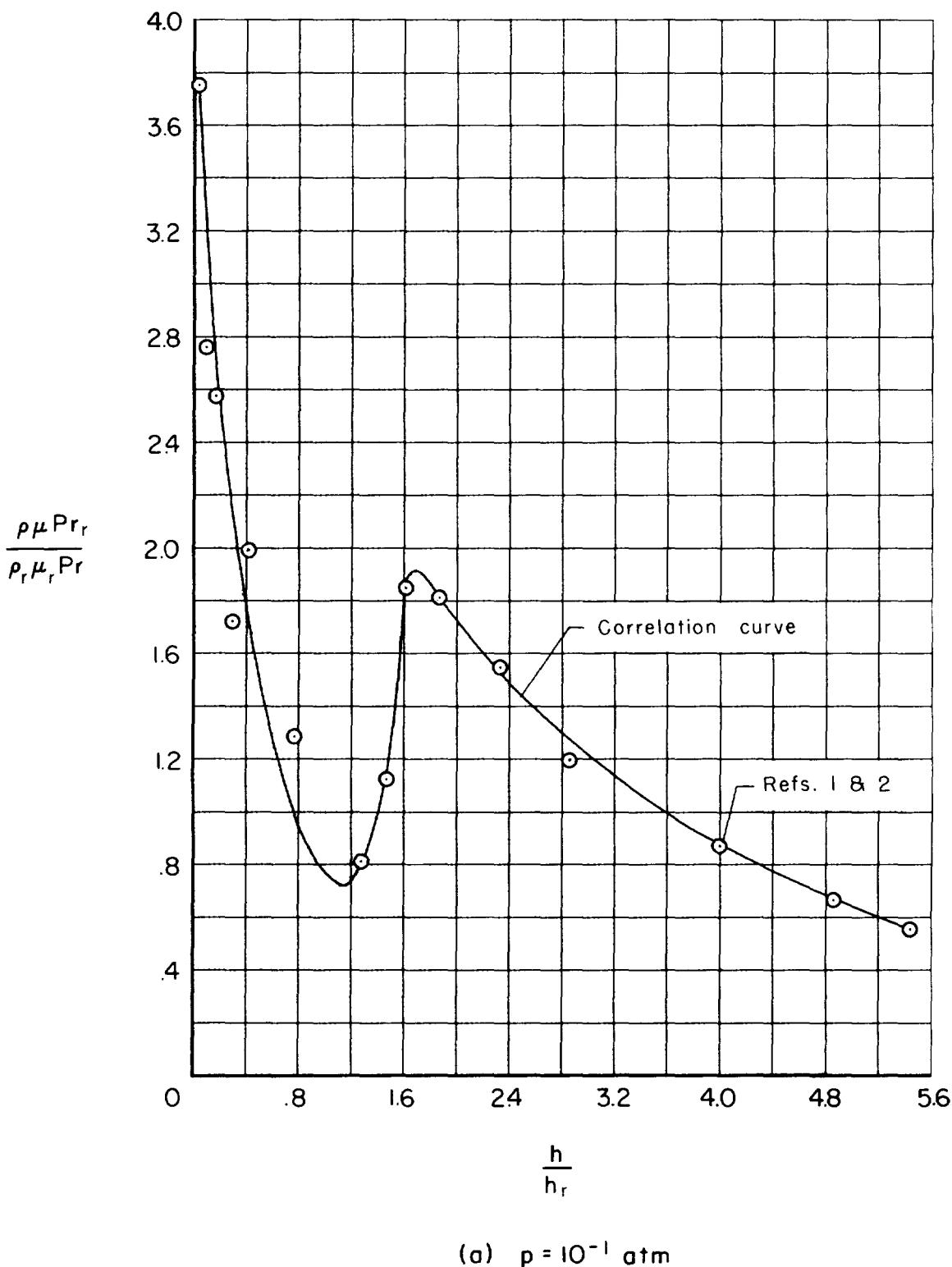


Figure 3.- Density-viscosity Prandtl number parameter correlation.

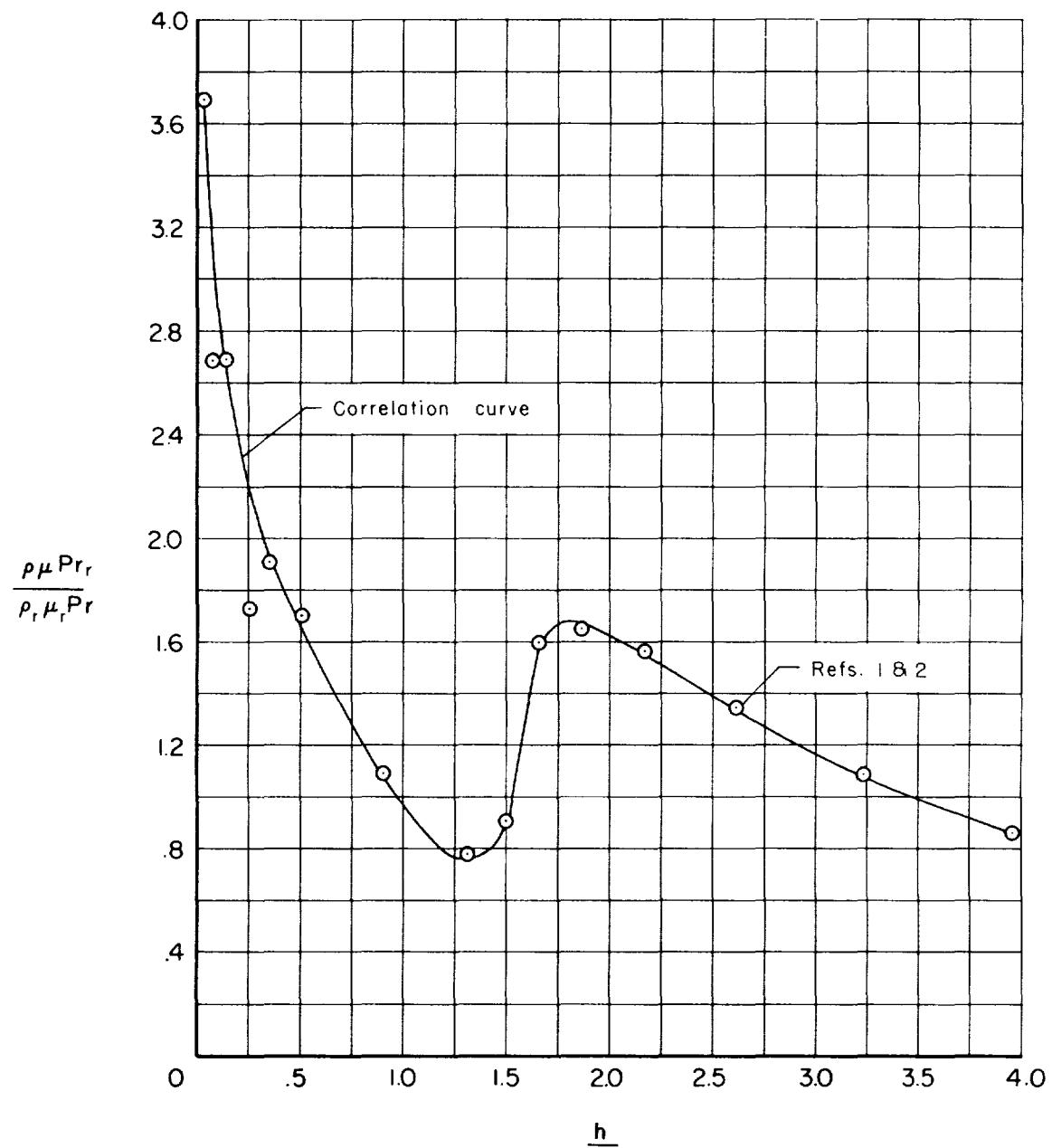
(b) $p = 1 \text{ atm}$

Figure 3..- Continued.

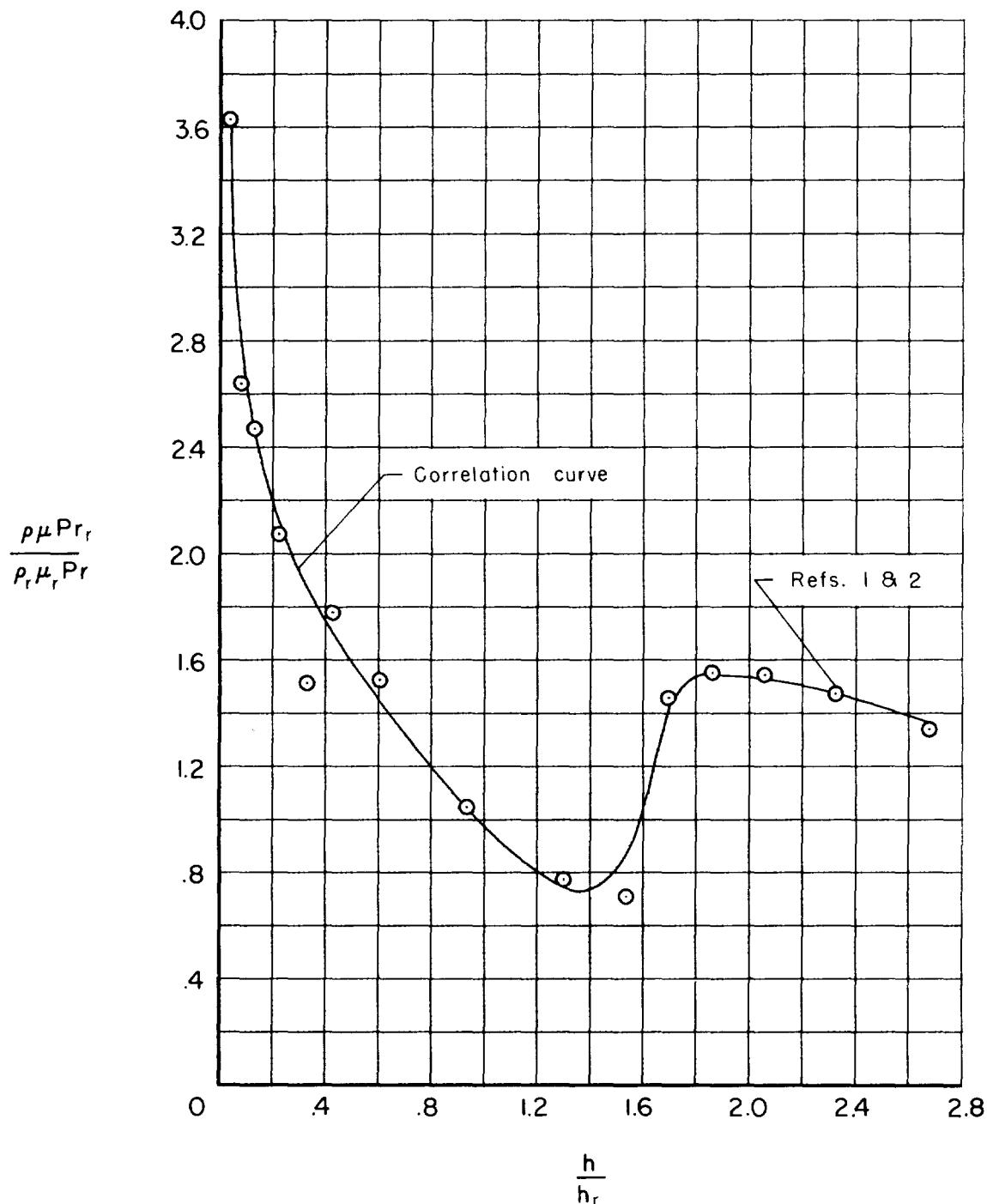
(c) $p = 10$ atm

Figure 3.- Continued.

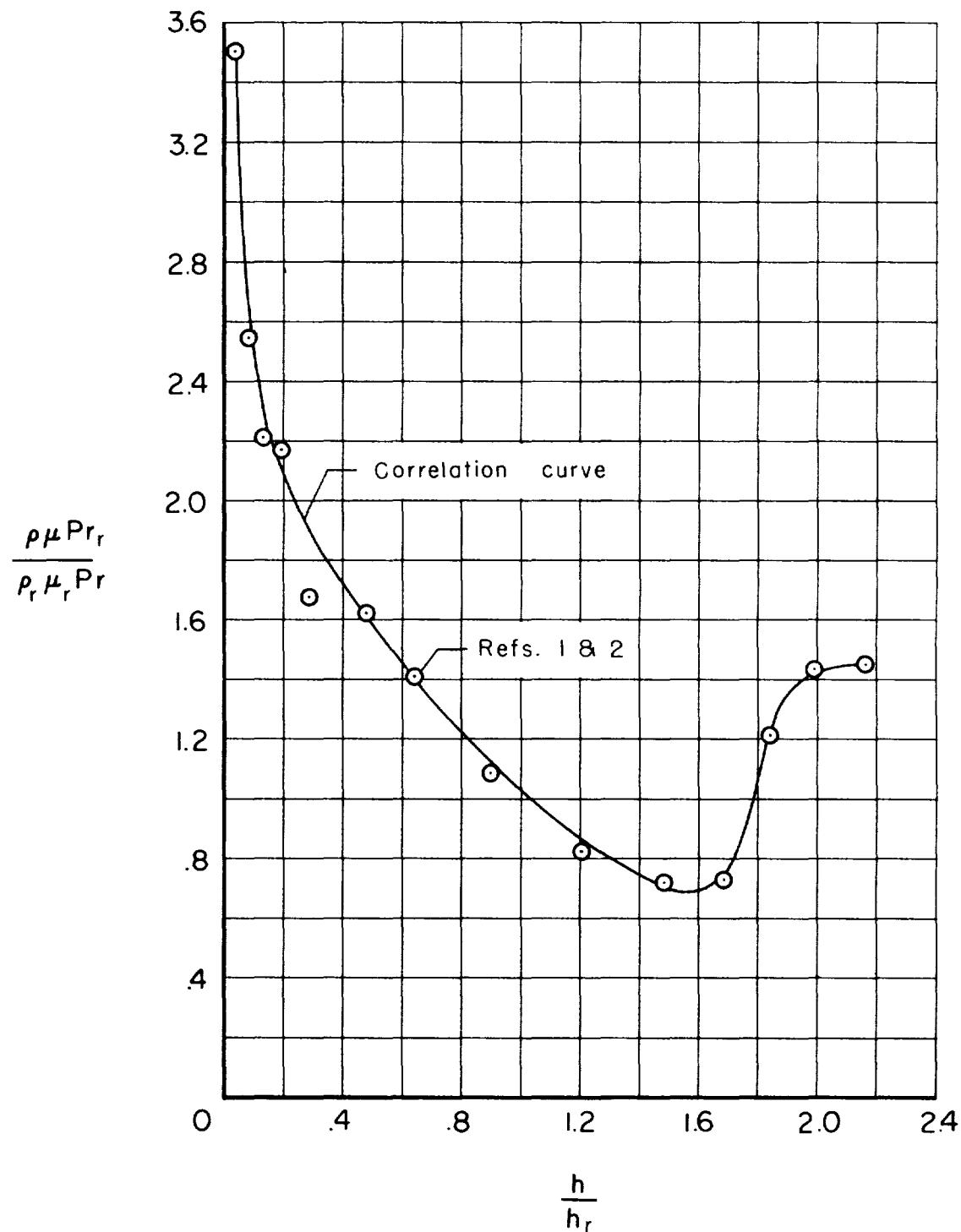
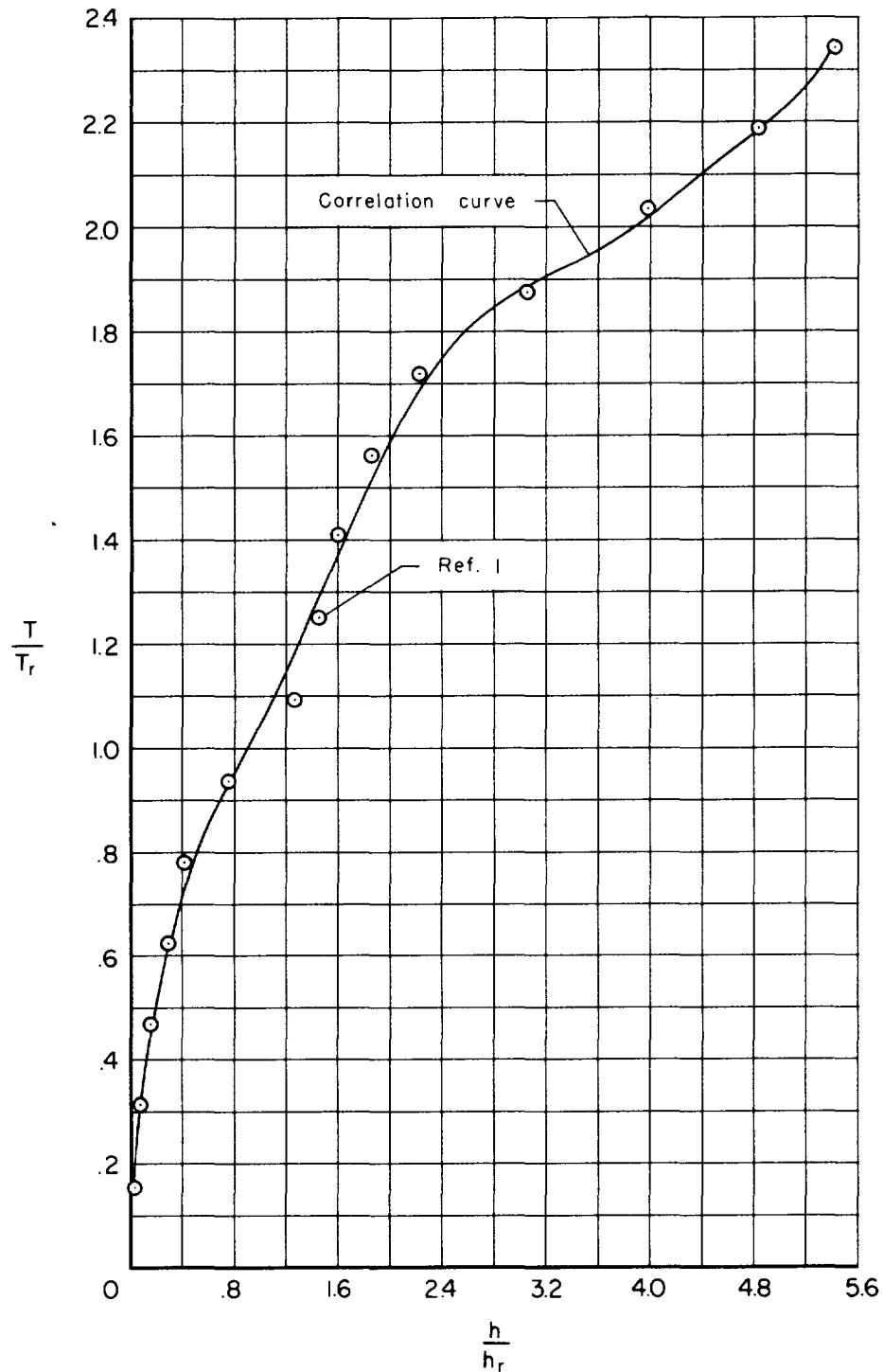
(d) $p = 100 \text{ atm}$

Figure 3.- Concluded.



(a) $p = 10^{-1} \text{ atm}$

Figure 4.- Temperature correlation.

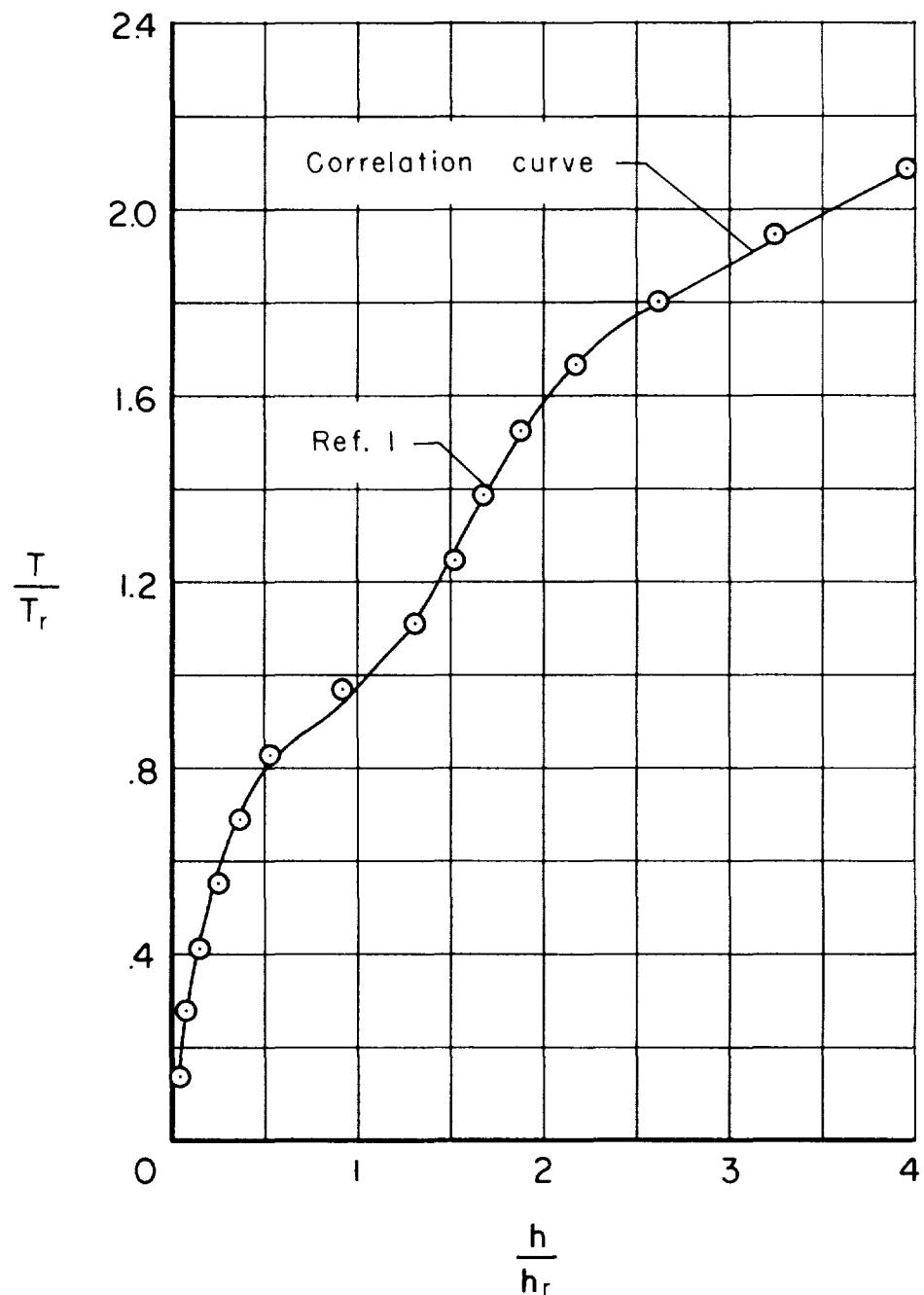
(b) $p = 1 \text{ atm}$

Figure 4.- Continued.

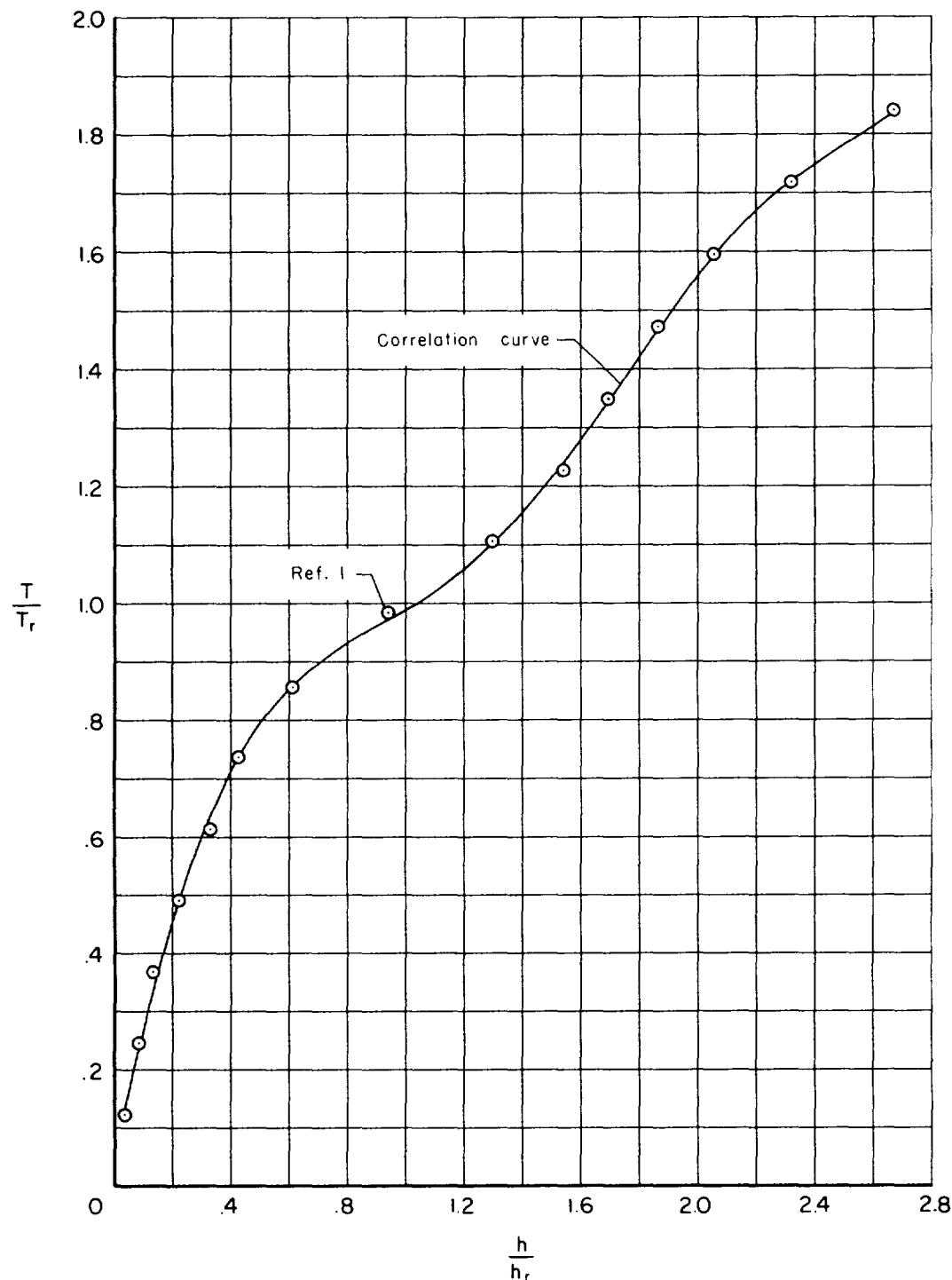
(c) $p = 10 \text{ atm}$

Figure 4.- Continued.

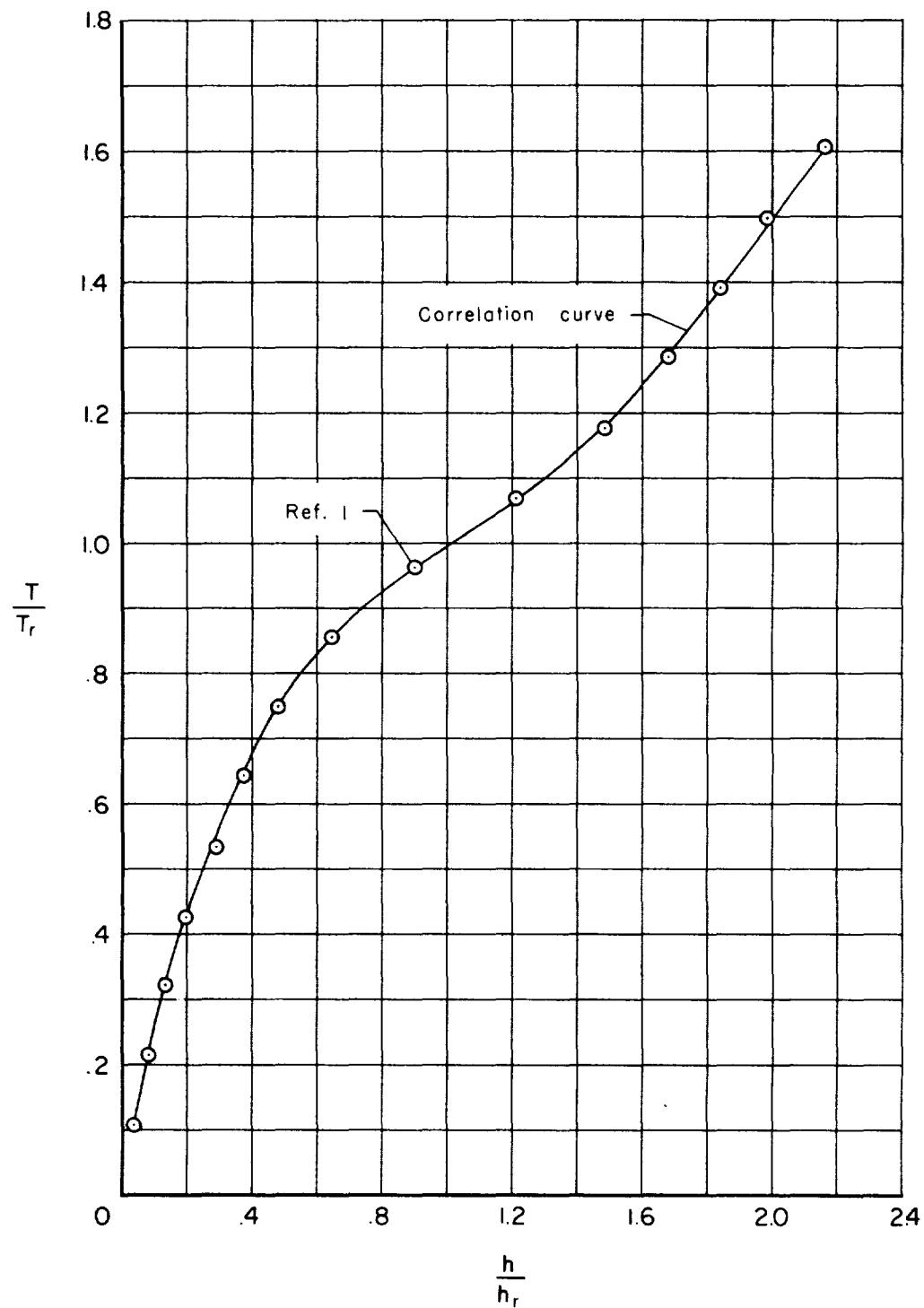
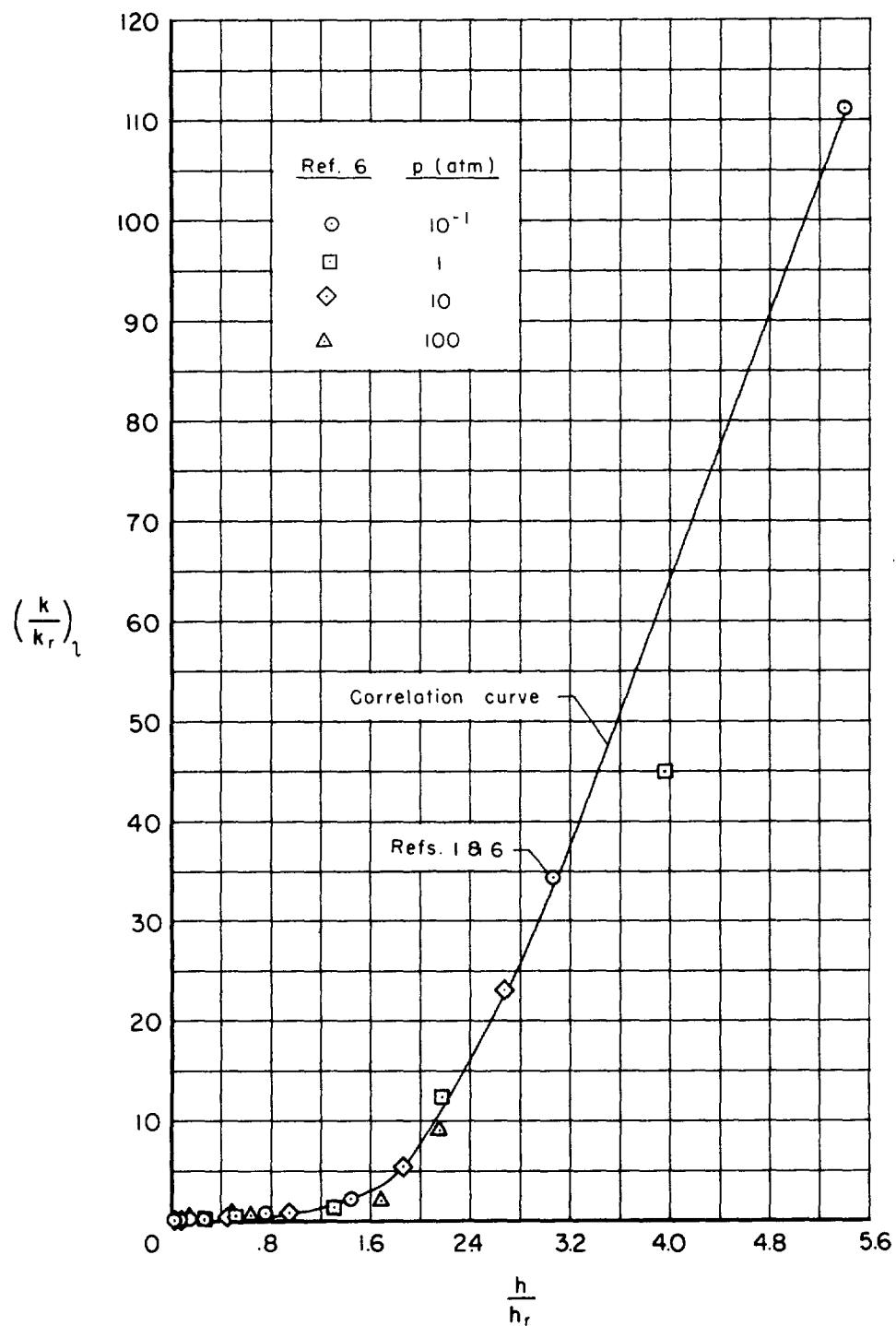
(d) $p = 100 \text{ atm}$

Figure 4.- Concluded.



(a) All pressures.

Figure 5.- Planck mean-mass absorption coefficient correlation (ref. 6).

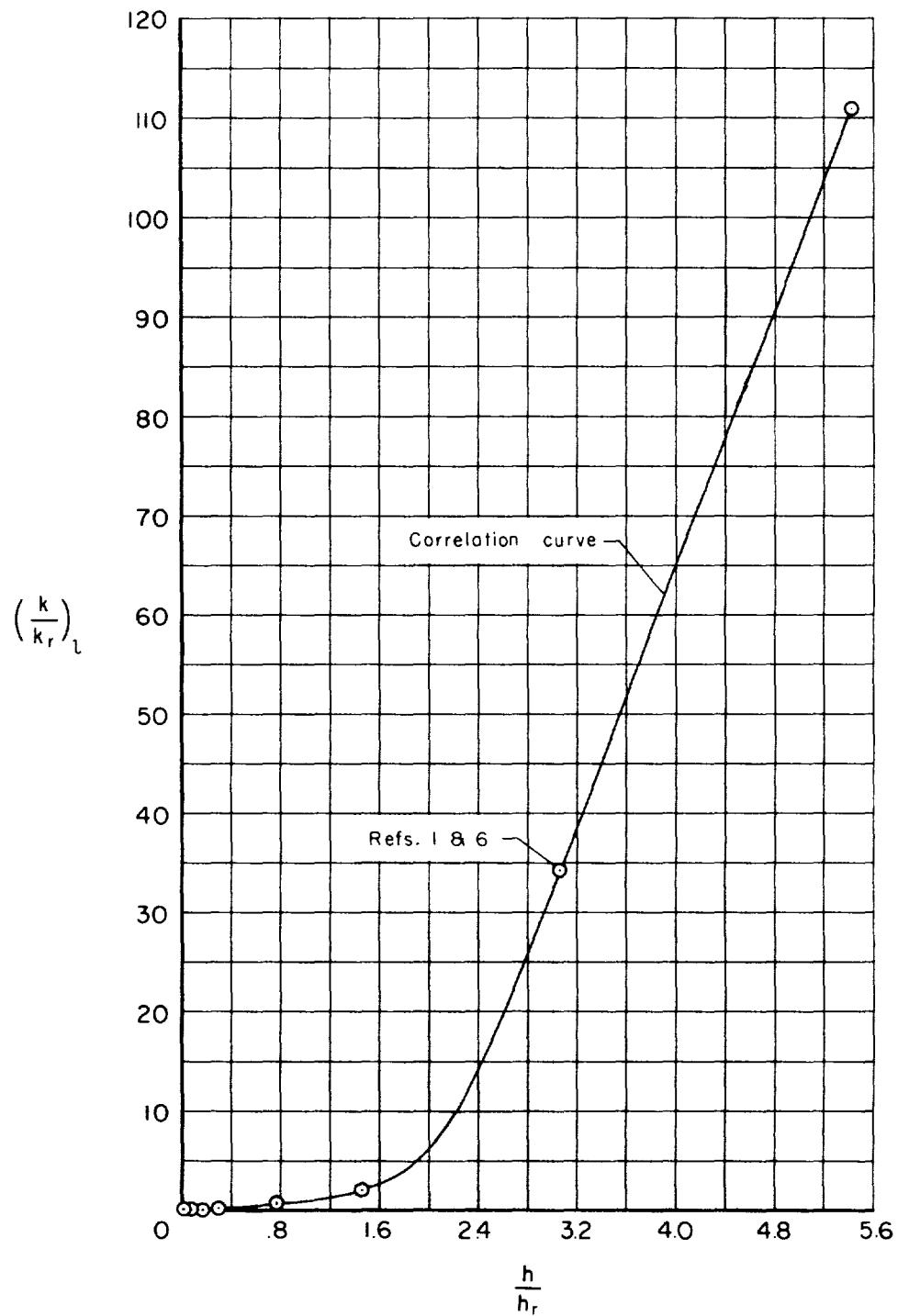
(b) $p = 10^{-1}$ atm

Figure 5.- Continued.

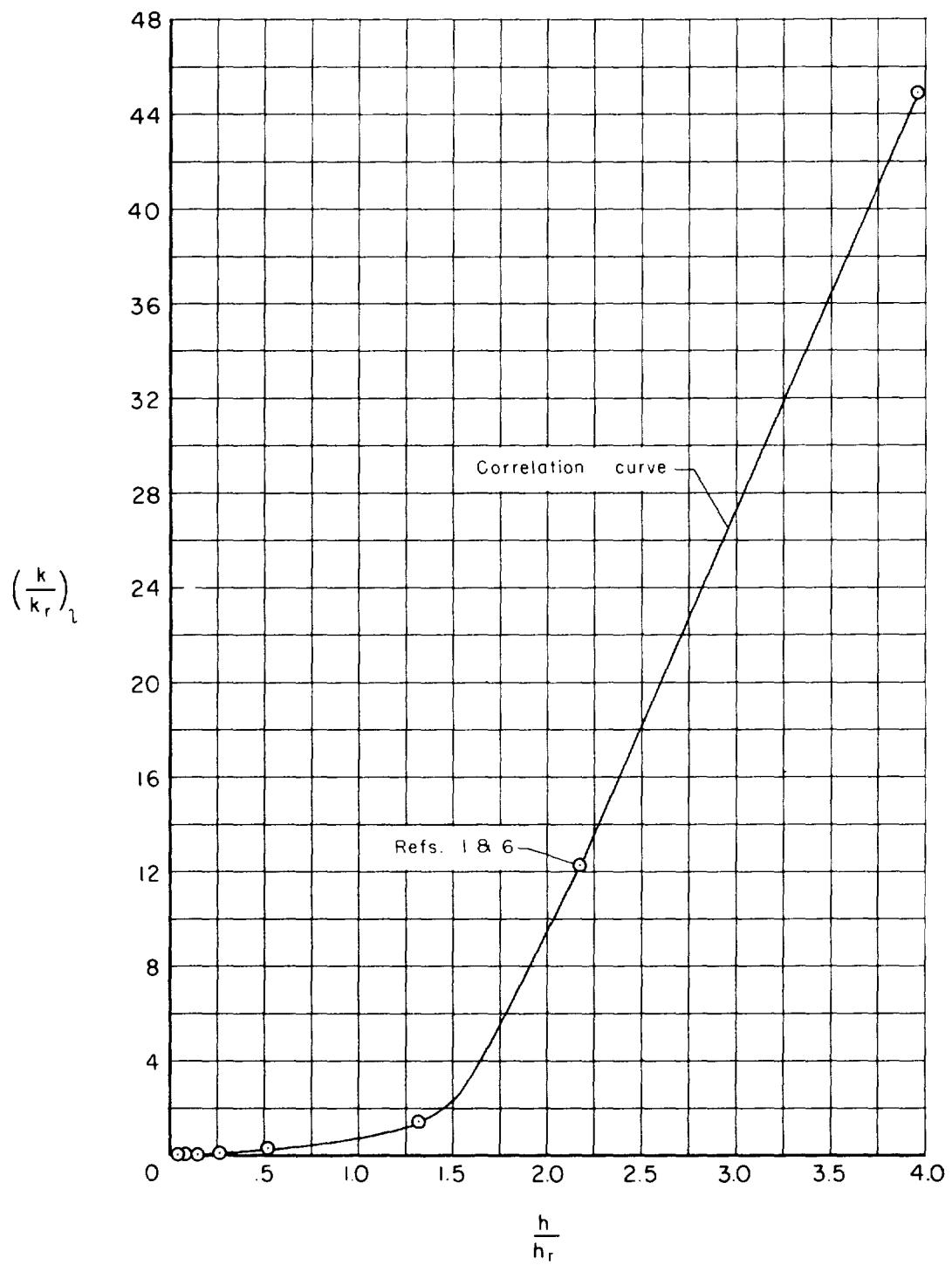
(c) $p = 1 \text{ atm}$

Figure 5.- Continued.

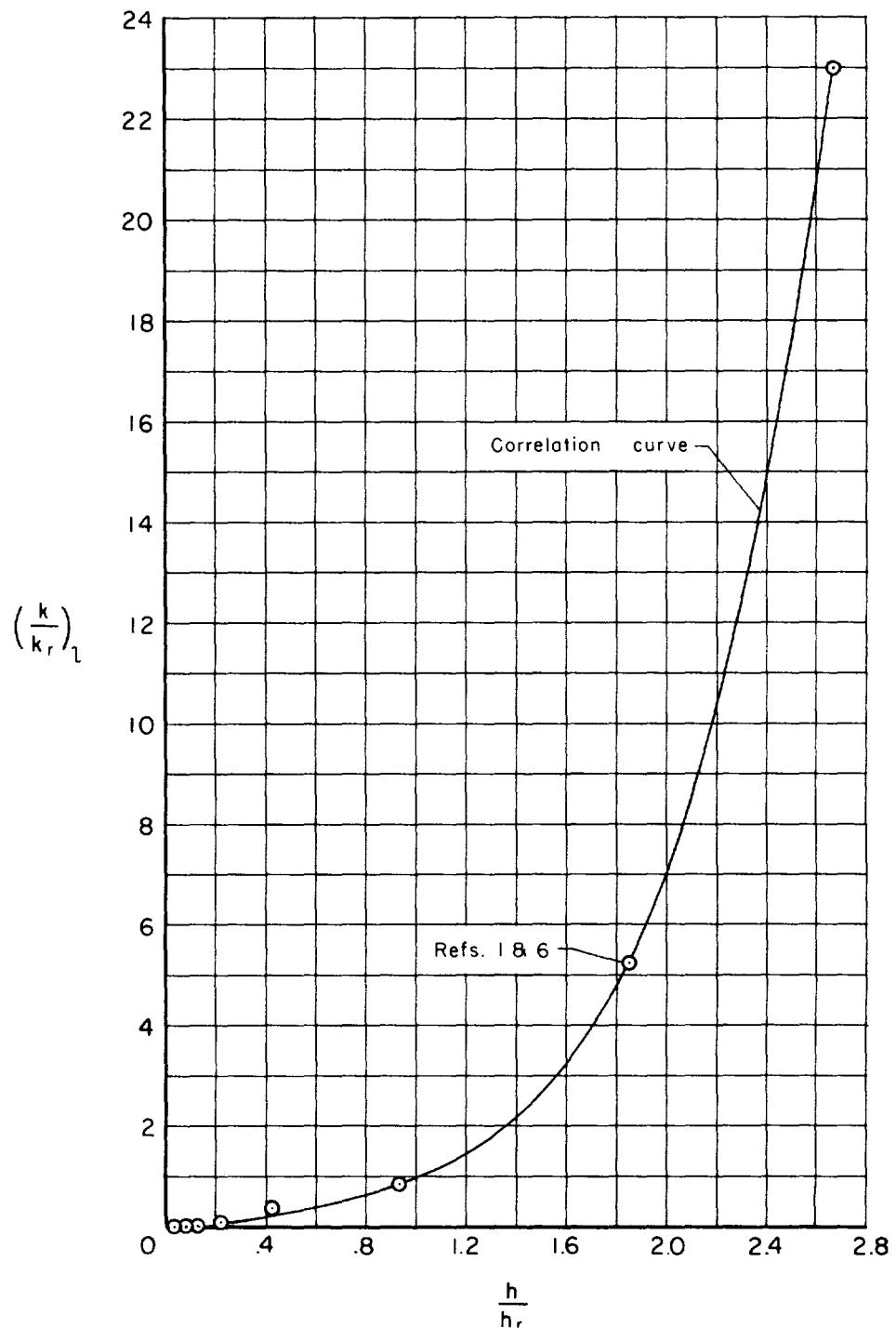
(d) $p = 10 \text{ atm}$

Figure 5.- Continued.

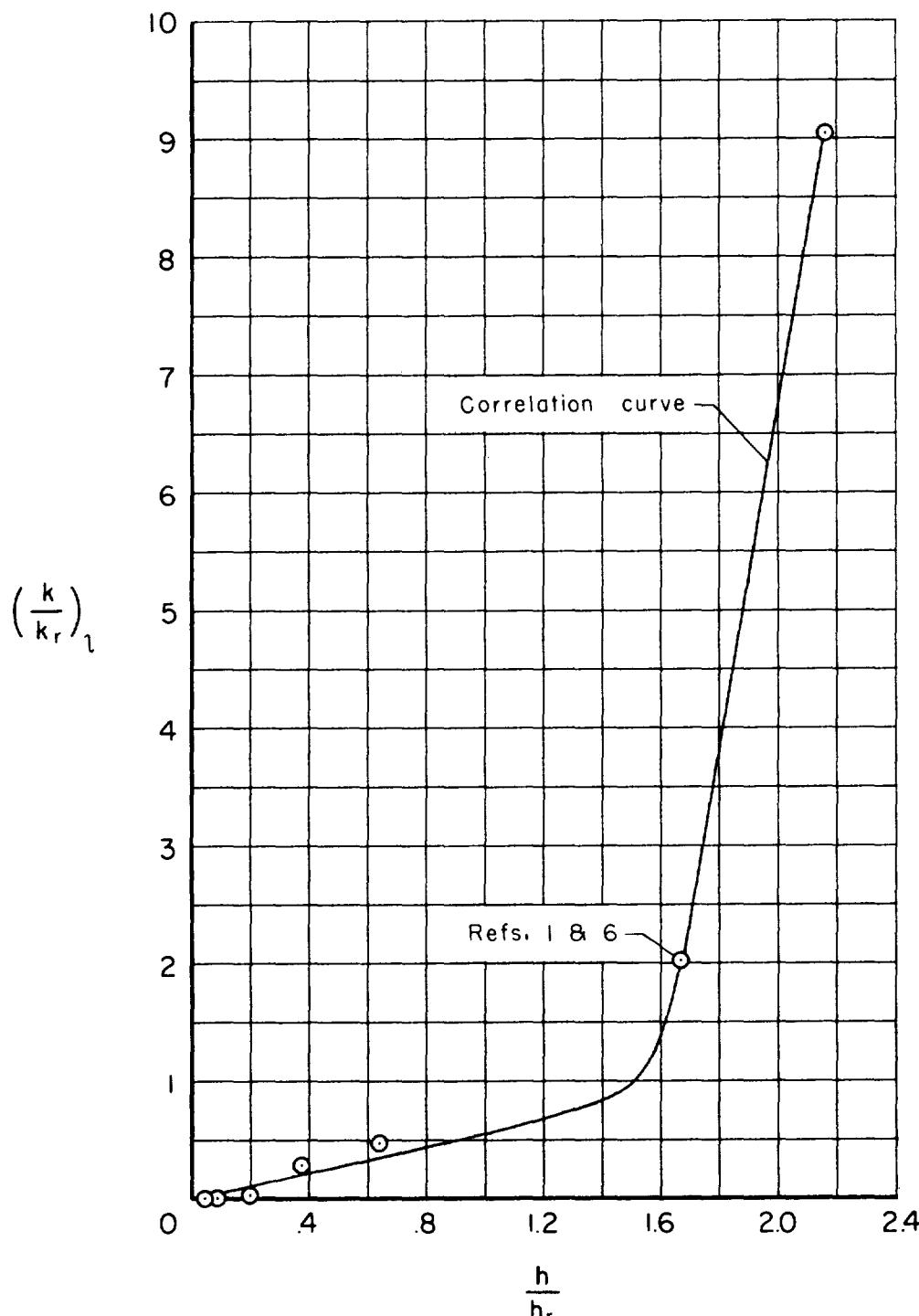
(e) $p = 100$ atm

Figure 5.- Continued.

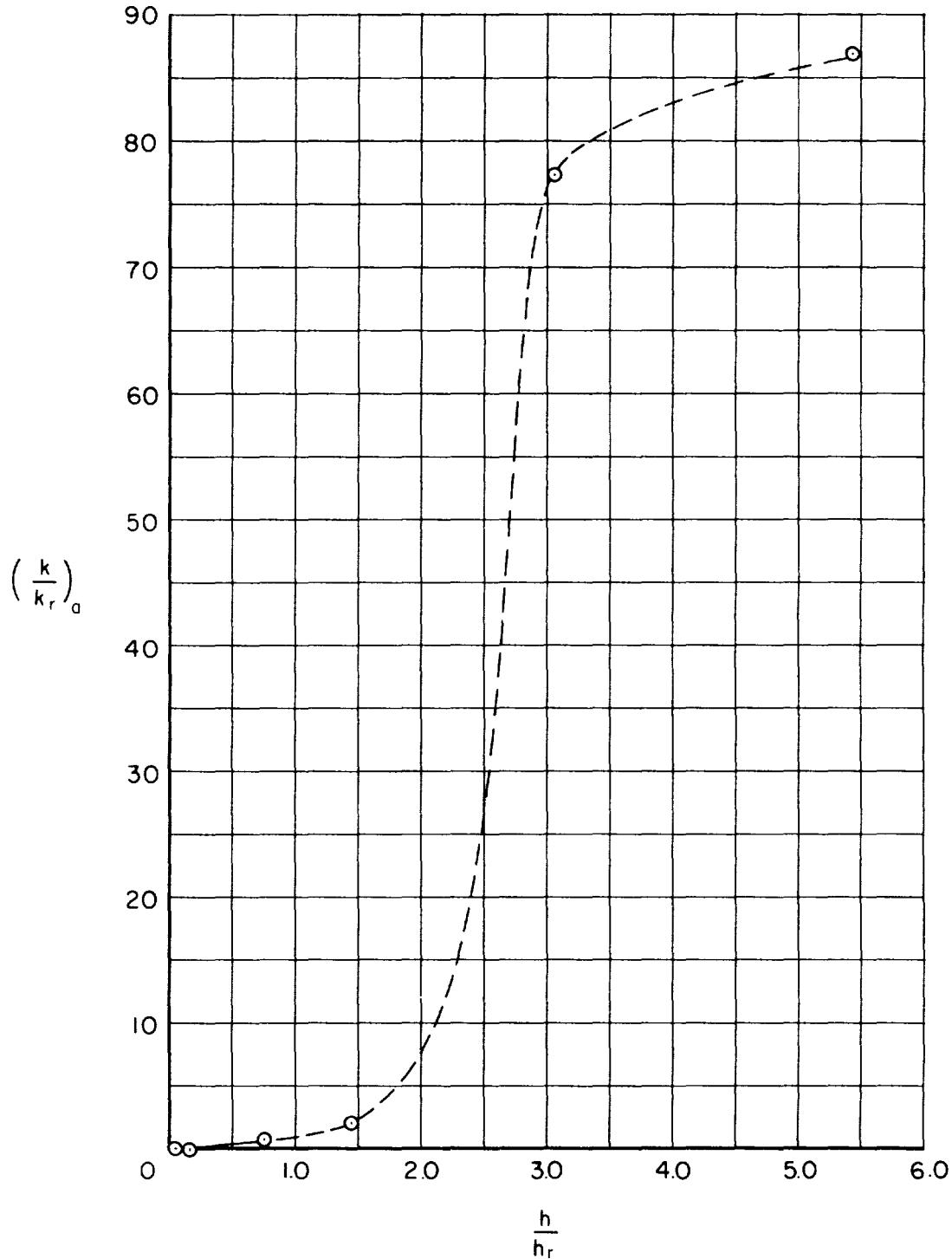
(f) $p = 10^{-1}$ atm

Figure 5.- Continued.

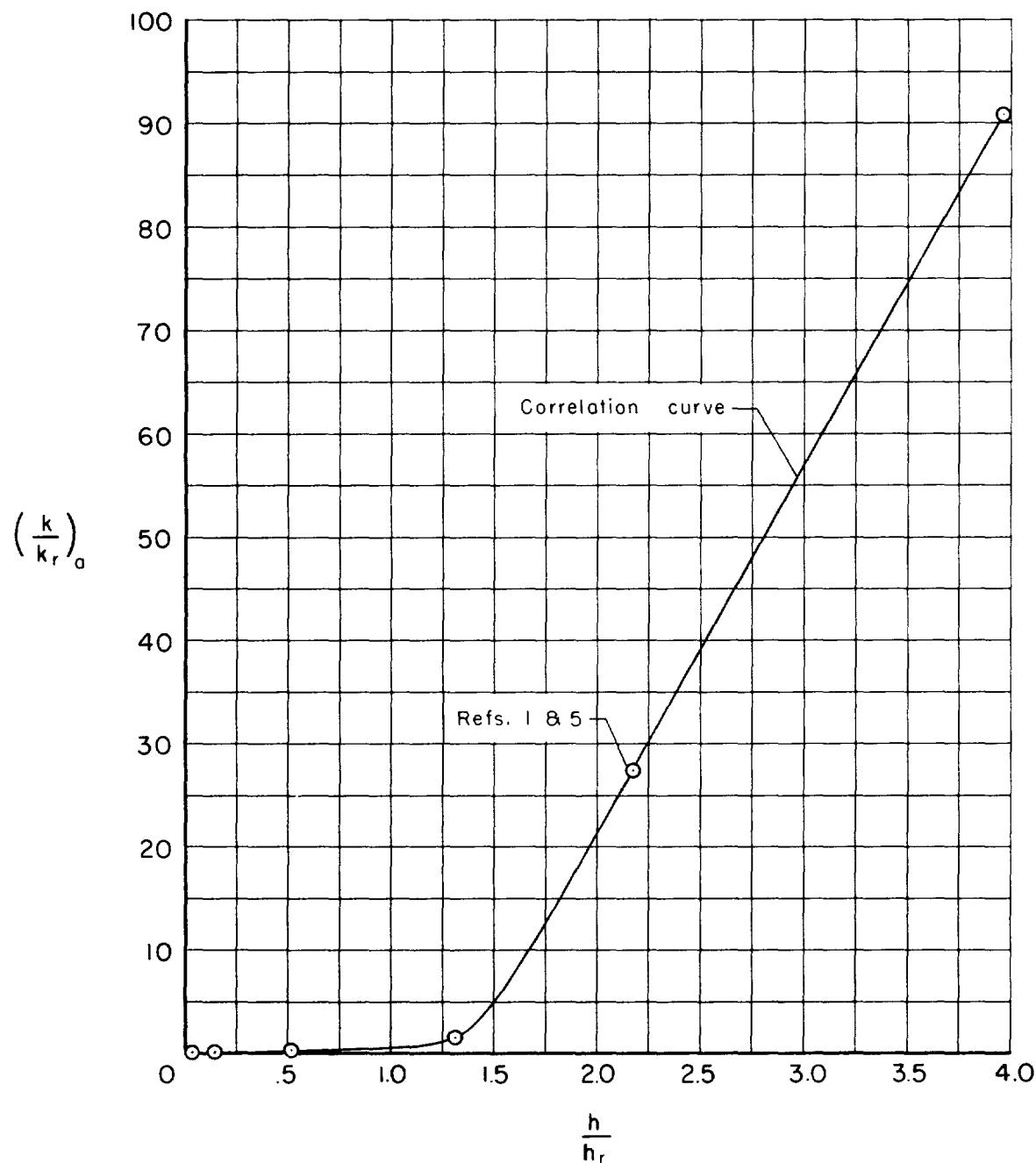
(g) $p = 1$ atm

Figure 5.- Continued.

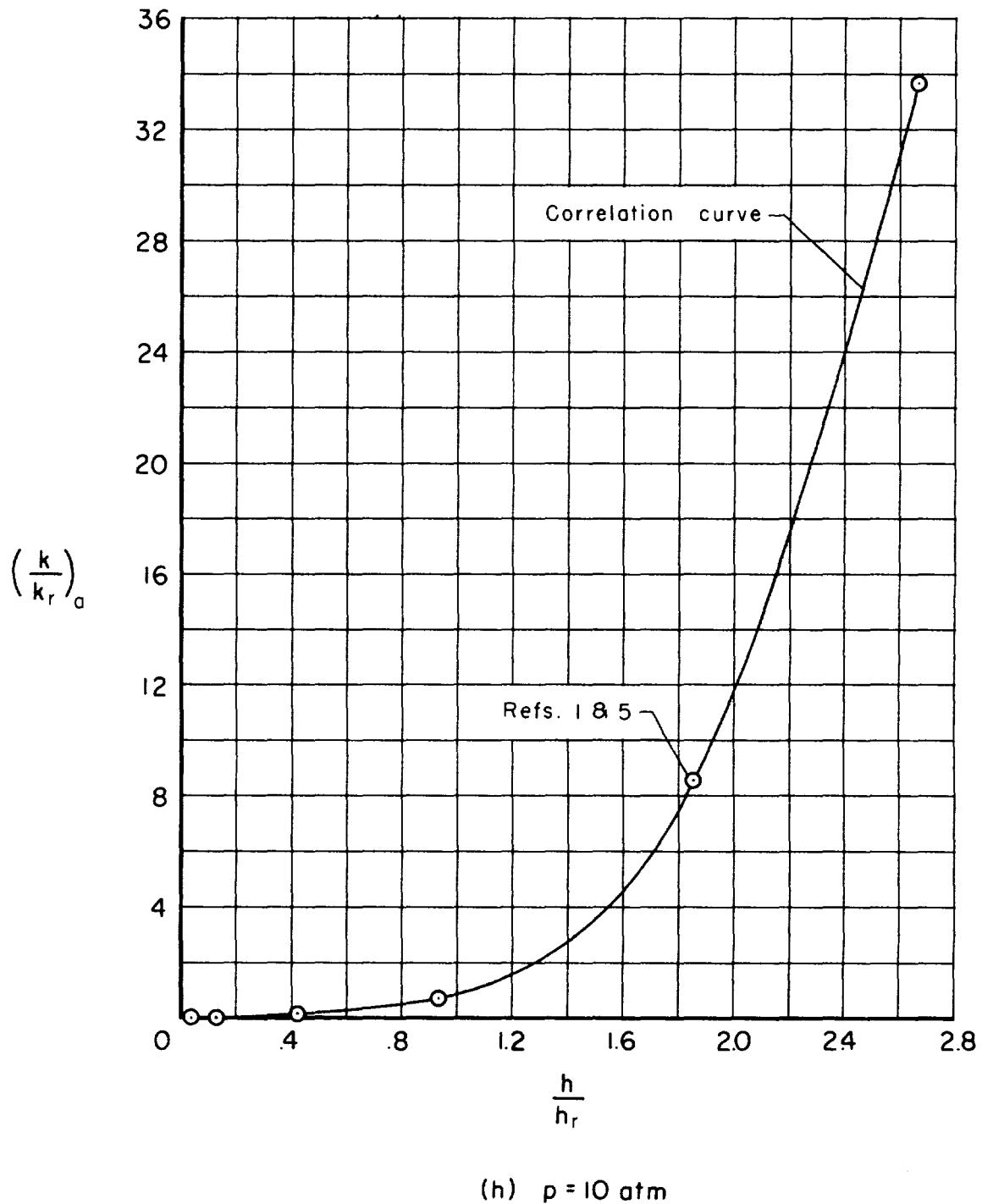


Figure 5.- Continued.

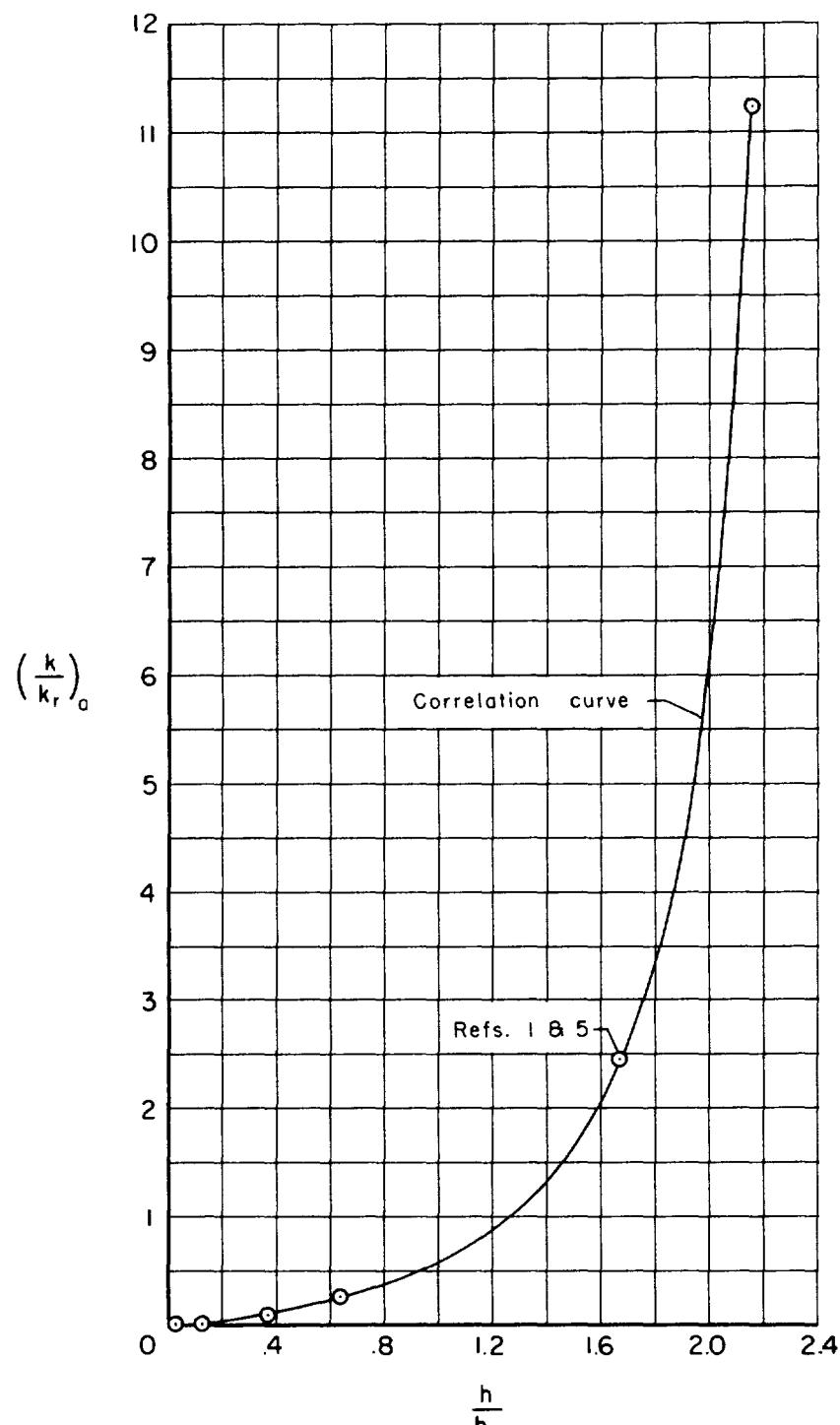
(i) $p = 100$ atm

Figure 5.- Concluded.

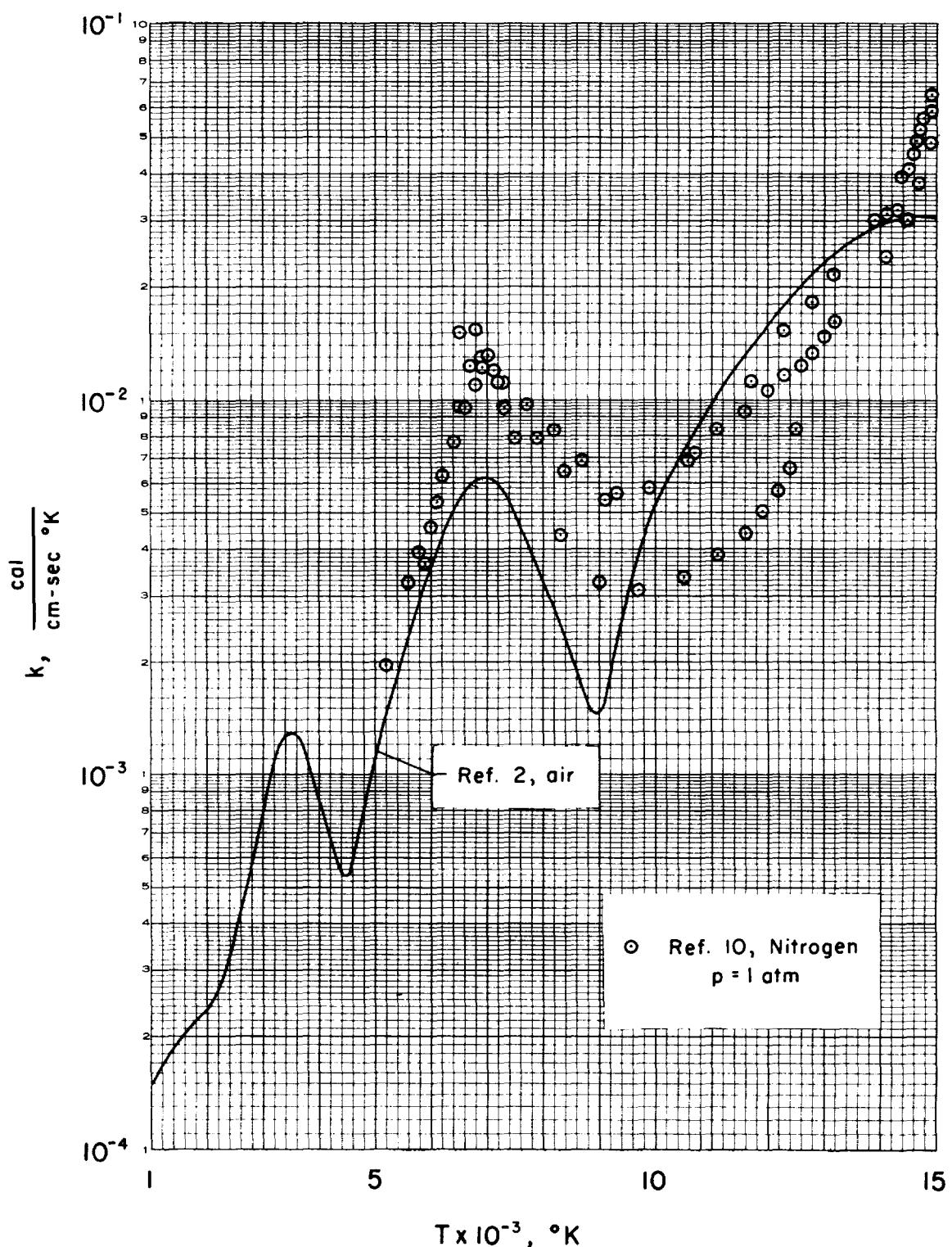


Figure 6.- Thermal conductivity - comparison of experiment and theory.

NASA-Langley, 1962 A-664

